

Dynamic characterization of timber floors sub-assemblies: sensitivity analysis and modelling issues

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ABSTRACT

Timber floors are prone to exhibit vibration levels which can cause discomfort to the occupants. In the last twenty years, ambient vibration tests have become very popular due to the many advantages they have over traditional forced vibration tests, when dealing with civil engineering structures. Furthermore, sensitivity analyses and "black box" optimization algorithms can support the development of refined finite element models that accurately predict the structures' responses based on the experimental modal parameters. However, applications of these methods and techniques to timber structures are scarce compared to traditional materials. This paper presents and discusses the findings of an experimental testing campaign on a lightweight timber floor. At first, each component of the assembly was tested separately under different boundary conditions. Then

24 the authors evaluated the behaviour of the whole floor assembly. In a second step, the authors car-
25 ried out a covariance-based sensitivity analysis of FE models representative of the tested structures
26 by varying the different members' mechanical properties. The results of the sensitivity analysis
27 highlighted the most influential parameters and supported the comparison between diverse FE mod-
28 els. As expected, the longitudinal modulus of elasticity is the most critical parameter, although the
29 results are very dependent on the boundary conditions. Then automatic modal updating algorithms
30 tuned the numerical model to test results. As a concluding remark, the experimental and numerical
31 results were compared to the outcomes of a simplified analytical approach for the floor's first natural
32 frequency estimate based on Eurocode 5.

33 INTRODUCTION

34 Modal testing represents a standard practice in structural engineering. Traditional modal testing
35 is based on estimating frequency response functions, which basically are the ratio of the output
36 response to the input excitation. This approach is also known as Experimental Modal Analysis
37 (EMA). Other ways to obtain modal properties through testing are the so-called Operational Modal
38 Analysis (OMA) methods. These approaches are very advantageous in civil engineering, where
39 the tested object is usually massive.

40 OMA encouraged copious research activities, which spanned from theoretical investigations
41 (Aloisio et al. 2020e; Reynders et al. 2012; Reynders et al. 2016) to practical applications (Bedon
42 and Morassi 2014; Rainieri et al. 2019; Aloisio et al. 2020a; Aloisio et al. 2020c). The scientific
43 literature documents a considerable amount of applications to civil engineering structures: wind
44 turbines (Tcherniak et al. 2011; Devriendt et al. 2014), stadiums (Peeters et al. 2007; Magalhães
45 et al. 2008), dams (Sevim et al. 2011; Pereira et al. 2018), architectural heritages (Kita et al. 2019;
46 Gentile et al. 2019; Antonacci et al. 2020; Aloisio et al. 2020d). The modal features, obtained
47 from OMA, bestow a direct insight into the actual structural behaviour and can guide a heedful
48 assessment about the modelling of the tested structures. A high-quality experimental campaign
49 can yield a reliable estimation of many modal parameters, valuable in understanding the limits and
50 advantages of the possible modelling approaches. The matching between the experimental modal

51 parameters and those obtained from the numerical model endorses the modelling choices. The
52 search for an optimum matching leads to an optimum model, obtained by optimizing the modelling
53 variables via the so-called model updating methods (Friswell and Mottershead 2013). Model up-
54 dating defines the process of refreshing the modelling variables at each step to minimize a proper
55 objective function, which magnifies the difference between experimental and numerical features.
56 In the digital era, model updating is gaining popularity due to automated optimization algorithms.
57 These algorithms lead to an optimum structural model, which best mirrors the experimental re-
58 sponse. The increasing popularity of model updating methods has alimeted considerable research.
59 Today, a researcher can use numerous optimization algorithms, which are equally feasible in terms
60 of reliability and computational efforts.

61
62 Timber is an excellent construction material with good stiffness-to-mass ratios and carbon-
63 storing properties. These characteristics has made timber very popular in the last years. The
64 interest in timber structures has risen, especially in the last two decades, due to the advent of new
65 engineering wood products, like the Cross-Laminated Timber (Ceccotti et al. 2013; Brandner et al.
66 2016; Izzi et al. 2018; Aloisio et al. 2020b). The low weight of timber, however, is a double-edged
67 sword to the dynamic performance. The use of timber elements is beneficial in reducing dead
68 loads (and inertial forces) on the structure. On the other hand, its low mass makes it prone to reach
69 a higher amplitude of vibrations. The assessment of timber buildings' vibration performances
70 has two primary branches: one focused on evaluating the lateral response(Reynolds et al. 2016;
71 Mugabo et al. 2019; Aloisio et al. 2020f; Aloisio et al. 2021), the other on assessing walk-induced
72 vibrations and the comfort requirements for the users (Smith et al. 2007). While the first field
73 is relatively new, researchers have investigated the second aspect for many years (Ohlsson 1982;
74 Smith and Chui 1988; Hu et al. 2001; Hamm et al. 2010). The serviceability limit state is related
75 to the perception of annoying oscillations caused by walking-induced vibrations. The "live" feel
76 of timber floors is familiar to many, especially in single-family housing with a timber framework.
77 However, this problem is not limited to timber-framed residential buildings. Timber joists can

78 support the flooring system even in masonry buildings (Hu et al. 2001). The trend of seeking
79 large, open-spaced architectural layout and adopting new construction practices certainly affects
80 timber floors' serviceability significantly. The ability to predict timber flooring systems' behaviour
81 remains a difficult task and a topical subject.

82 There are some applications of ambient vibration tests on timber floors in the scientific liter-
83 ature (Weckendorf and Smith 2012; Weckendorf et al. 2014; Weckendorf et al. 2016). However,
84 force vibration tests, and EMA methods remain the most known and used procedures to estimate
85 traditional floors' modal properties or more innovative solutions (e.g. CLT and Timber concrete
86 composites) (Casagrande et al. 2018; Xie et al. 2020; Huang et al. 2020). Applications of OMA
87 methods and automated modal updating procedures to timber structures are still not copious. This
88 paper presents and discusses ambient vibration test results of a timber floor and the modelling
89 strategies and techniques adopted to simulate the floor's dynamic response numerically. Specifi-
90 cally, the research studies the response of two glulam beams with plywood decking, which are part
91 of a simply-supported timber floor. At first, each assembly component was tested separately under
92 different boundary conditions; then, the authors evaluated the whole floor assembly's behaviour.
93 In a second step, the authors carried out a covariance-based sensitivity analysis on the FE models
94 representing the tested structures by varying the Moduli of Elasticity of the different members. The
95 sensitivity analysis outcomes evidence the significant structural parameter and drive a definitive
96 comparison between diverse FE modelling methods. The authors used two automated updating
97 algorithms to refine the numerical model's parameters better and match the testing results. The
98 adoption of closed-form analytical solutions is diffuse in engineering practice. Therefore, the
99 authors compared the well-known Euler-Bernoulli model for the simply-supported beam tests with
100 the FE numerical predictions.

101 **MATERIALS AND METHODS**

102 The authors tested a timber floor sub-assembly made by two beams and decking above. The
103 two GL30C beams are 5m long with a 115mm x 315mm cross-section. The nominal average
104 Modulus of Elasticity (MoE) is $13GPa$, while the mean weight is $430kg/m^3$, according to EN

105 14080 (EN14080 2013). Both beams presented some defect at delivery, see Fig1. "Beam 1" on
106 one end had two cracks (approximately 15cm and 20cm wide), on both faces; while "Beam 2" had
107 a hole on one face that was filled with silicone.

108 The decking consists of 21 mm thick Plywood 1,5 x 1,5 m sheets made from Birch veneers.
109 According to the producer declaration of performance (DoP) the self-weight is $650\text{kg}/\text{m}^3$, while
110 the mean values of the MoE span between 6GPa and 8GPa , depending on loading direction,
111 perpendicular or parallel to the external layer fibre orientation, respectively.

112 In modal testing practice, mechanical parts, machinery and other structural components are
113 tested freely-suspended due to the difficulties in modelling the boundary conditions. Due to the
114 laboratory conditions, it was not possible to suspend the beams. The authors adopted a compromise
115 solution, based on the use of a layer of Rockwool insulation placed under the beams, which
116 successfully simulated the free-free boundary conditions. A single rectangular piece of Rockwool
117 (300mm x 300mm, 100mm thick) located under the mid-span of the beams, or by the centre of
118 the plywood boards yielded the best results, in terms of repeatability, consistency and clearness of
119 both the spectral densities and the stabilisation diagrams. The presence of the Rockwool layer may
120 affect the results in terms of damping. However, reliable damping estimates are always challenging
121 to achieve and are not the primary scope of this investigation.

122 Pinned-pinned boundary conditions characterized the floor assembly in Fig2. Two metal
123 cylinders, spaced 4.8m, supported each beam, with a 600mm centre-to-centre distance. The
124 decking was made of three square boards with 1.5m long sides. The beams, being 5m long, were
125 not covered by the boards for the last 25cm on each side, see Fig2a. Furthermore, no nailed or
126 screwed connector secured the boards over the beams. Dynamic analyses are susceptible to the
127 occurrence of little damage or minimal structural modification. The insertion of the connectors
128 would have altered/damaged each component, thus nullify the efforts to identify the dynamics
129 of each of them accurately. Therefore, the authors devised an alternative solution to study the
130 entire structural arrangement without the need for connectors. They placed a reusable putty-like
131 pressure-sensitive adhesive, which guarantees the joint response of the beams and the decking in

132 the vertical direction. Even if in a real building the decking would be fixed to the beams, thus
133 enhancing the composite interaction and the overall stiffness, the floor would also be much thicker
134 and heavier, due to the finishing. The structural assembly is not intended to be representative
135 of realistic situations, it is a structural archetype useful for the accurate calibration of numerical
136 models able to predict its vibration performance.

137 A slight and random brushing of the structures using a wooden stick represented the excitation
138 source. This method aims to improve the signal-to-noise ratio of the measurements (Brincker and
139 Ventura 2015). The Enhanced Frequency Domain Decomposition method (EFDD, (Brincker et al.
140 2001)) and the Stochastic Subspace Identification method (SSI-cov (Peeters and De Roeck 1999),
141 SSI-dat (Van Overschee and De Moor 2012)), implemented by the authors in Python programming
142 language, yielded the modal parameters from the acquired data for the wooden beams and decking
143 under investigation. The EFDD method, which is a so-called non-parametric, frequency domain
144 procedure, and SSI, which is a parametric, time-domain procedure, are probably among the two
145 most used techniques for OMA.

146 The numerical characterisation of the dynamic response originated from Finite Element Mod-
147 elling using the software SAP2000 (CSI 2020). The authors developed a set of models for each
148 sub-assembly (i.e. beams and board) before the testing using standardised values for the material
149 properties (i.e. from material standard and DoP). These models provided an expected response,
150 which was useful to derive a proper setup and instrumentation plan. Two models reproduced the
151 dynamics of the beams. The former derived from the one-dimensional "Frame elements" based
152 on the Timoshenko beam theory, the latter originated from the use of "Solid elements", which are
153 eight-node elements for modelling three-dimensional structures. The material property was defined
154 as orthotropic to model the glulam. Thin "Shell elements" modelled the decking, with the plywood
155 of the boards idealised as an orthotropic material. Unfortunately, SAP2000 does not perform a
156 modal analysis of unrestrained objects. Therefore, a "Linear-link" element connected the modelled
157 structures' end corners to the ground. An infinitesimal stiffness was assigned to the link elements
158 to simulate the unrestrained boundary conditions.

159 The global model of the floor emerged from the sensitivity analysis and model updating of the
160 structural sub-assemblies, see Fig3. "Linear-link" elements connect the beam's nodes to the nodes
161 of the plywood boards. Each element is assumed to be composed of six separate "springs", each
162 associated with a deformational degree of freedom (DoF). Given the type and source of loading,
163 the authors assigned an infinite stiffness to the first local axis of the spring, representing the contact
164 between components (see Fig3). Conversely, the other DoF were kept unrestrained since the boards
165 were not fixed to the beams. The mesh size of the frame elements (50 mm), the solid elements
166 (55x25x30 mm) and the shell elements (50x50 mm) derived from a simple convergence test on the
167 firsts natural frequencies, and represent a possibly satisfactory compromise between accuracy and
168 computational time.

169 The SAP2000 Open Application Programming Interface (OAPI) was used in combination with
170 the open-source programming language Python to develop the routines for the sensitivity analysis
171 and model updating. The OAPI allows third-party products, like Python, to interact with SAP2000,
172 allowing the users to create custom applications.

173 A Sobol sensitivity analysis (Sobol 1993) evidenced the role of each term of the flexibility
174 matrix of an orthotropic finite element. Namely, the analysis returned the sensitivity indices of the
175 three MoE, E_X E_Y E_Z , the three Shear Moduli, G_{XY} G_{XZ} G_{YZ} , and three Poisson's ratios, ν_{YX} ν_{ZX}
176 ν_{ZY} on the output (modal properties).

177 Finally, the FE models were tuned to reflect the measured data better using two global opti-
178 mization algorithms for "black box" functions, the Differential Evolution (DE) (Storn and Price
179 1997) and the Particle Swarm Optimization (PSO)(Kennedy and Eberhart 1995). The script for
180 the model updating process was written in Python using SAP2000 OAPI along with the Python
181 module PySwarms (Miranda 2018) (to run PSO), and the popular Python toolkit SciPy (Virtanen
182 et al. 2020)(to run DE). The idea behind PSO is to emulate the social behaviour of birds and
183 fishes by initializing a set of candidate solutions to search for an optimum. A set of candidate
184 solutions (called particles) are moved around in the search-space. The movements of the particles
185 are guided by their own best-known position in the search-space as well as the entire swarm's

186 best-known position. Differential evolution is a stochastic population-based method that, at each
 187 step, mutates each candidate solution (called agents) by mixing with other candidate solutions to
 188 create a trial candidate. If the new position is an improvement, then it is accepted and forms part
 189 of the population. Otherwise, the new position is simply discarded.

190 The following objective function measures the distance between the estimated modal parameters
 191 and the numerical ones:

$$192 \quad C = \sum_{i=1}^M \gamma_i \left(\frac{f_i^m - f_i^c}{f_i^m} \right)^2 + \sum_{i=1}^M \beta_i (1 - \text{MAC}(\{\phi^m\}_i, \{\phi^c\}_i)) \quad (1)$$

193 where the apex $(*)^m$ indicates a measured variable, the apex $(*)^c$ a calculated variable, f_i is the
 194 i^{th} natural frequency, ϕ_i is the mode shape vector, M is the number of modes, MAC is the Modal
 195 Assurance Criterion, while γ_i and β_i are weighting factors.

196 Practitioners usually rely on simplified equation provided by building codes and standards
 197 to design structural elements, rather than rely on cumbersome and time-consuming FE analysis,
 198 especially at early design stages. To reflect this aspect the authors drew some comparisons to
 199 well-known engineering procedures. The bending vibrations of a beam can be described by the
 200 well-known Euler-Bernoulli beam equation:

$$201 \quad EI \frac{\partial^4 z}{\partial x^4} + \rho A \frac{\partial^2 z}{\partial t^2} = 0 \quad \text{with} \quad 0 < x < L \quad (2)$$

202 where the E is the MoE, I is the second moment of inertia of the cross-section, ρ is the mass
 203 density (mass per unit length), A is the cross-section area, z is the vertical displacement, L is the
 204 length of the beam and t is time. The solution for Eq.(2) can be found for example by decomposing
 205 the displacement into a sum of harmonic vibrations $z(x, t) = \text{Re}[\hat{z}(x)e^{-i\omega t}]$. Eq.(2) can then be
 206 rewritten as an ordinary differential equation $EI\partial^4\hat{z}/\partial x^4 - \rho\omega^2\hat{z} = 0$, which have a general solution
 207 of the form:

$$208 \quad \hat{z}_n = C_1 \cosh(k_n x) + C_2 \sinh(k_n x) + C_3 \cos(k_n x) + C_4 \sin(k_n x) \quad \text{with} \quad k_n = \left(\frac{\rho \omega_n^2}{EI} \right)^{1/4} \quad (3)$$

209 where $C_1 - C_4$ are constants that depend on the boundary conditions, k_n is the wave number and
210 ω_n is the n^{th} natural frequency. Eurocode 5 (EN1995 2004) provides a formula to estimate the first
211 natural frequency of rectangular floor with span L , width B , simply supported along the four edges,
212 which derives from the temporal component of the solution of Eq.(3):

$$213 \quad f_1 = \frac{\pi}{2L^2} \sqrt{\frac{(EI)_L}{m}} \quad (4)$$

214 where $(EI)_L$ is the equivalent bending stiffness along the span direction and m is the mass per unit
215 floor area.

216
217 To sum-up the following steps were pursued after the dynamic testing of the elements:

- 218 • Sobol sensitivity analysis to find the most important mechanical parameters.
- 219 • Model updating to tune the numerical model to the experimental results.
- 220 • Comparison between experimental results, numerical model results and analytical model
221 results.

222 **DYNAMIC IDENTIFICATION**

223 **Experimental setup**

224 The measurement chain was composed of ten seismic ceramic shear piezoelectric accelerome-
225 ters, an HBM QuantumX data acquisition unit (24-bit analogue-to-digital converter) and a laptop
226 pc. Shielded polyurethane coaxial cables made the connection between the sensors and the acquisi-
227 tion unit. The accelerometers (PCB, model 393B12) have an approximate 10000 mV/g sensitivity,
228 a frequency range from 0.15 Hz to 1000 Hz and a measurement range up to $\approx \pm 5 \text{ m/s}^2$.

229 The accelerometers measured the beam responses parallel to the principal axes of inertia (strong
230 and weak) in the free-free condition, according to the setups shown in Fig4a. Mounting studs and
231 small metal plates screwed to the beams extrados (i.e. top surface) secured the accelerometers to
232 the elements. The second setup for the weak axis allowed the extraction of the torsional modes of
233 the beams, see Fig4c. In this case, three different measurements were processed and then merged to

234 get the mode shapes. The sensors were attached to the beams through adhesive rubber to fasten up
 235 the testing operations. Furthermore, the beams were also tested on two metal supports to simulate
 236 the simply-supported condition; this time, the measurement axes were parallel to the strong axis
 237 of inertia. Fig4b shows the test setup of the decking. The authors tested a single panel out of the
 238 three plywood sheets. The testing of the structural sub-assembly had the sensors placed by the
 239 intrados of the beams. This choice allowed to leave the space on top of the floor free, see Fig2b.
 240 The accelerometers were evenly distributed along both beams. The distance between the edge
 241 accelerometers was lesser than the beam length due to the presence of the supports.

242 The sampling frequency was set to 1200 Hz (the aliasing filter is automatically set by the
 243 software embedded in the logger), and the duration was 5 minutes for every test. The data were first
 244 detrended to remove the DC offset with the application of a digital high-pass filter, then decimated.
 245 Different decimation factors, depending on the frequency bandwidth of interest, were used.

246 With regards to the EFDD method, the Power Spectral Densities (PSD) were estimated according
 247 to the Welch's method, dividing the data so as to get a frequency resolution of 0.1 Hz and using a
 248 Hanning window with 50% overlap. The MAC rejection level to estimate the single-DoF PSD "bell"
 249 function was set to 0.95. Twenty consecutive peaks were used to estimate the damped frequency and
 250 the damping ratio from the autocorrelation function, ignoring the first 3. For the SSI-cov method,
 251 the number of block rows was set to 15, and the maximum model order to 80. As suggested in
 252 (Rainieri and Fabbrocino 2014), the stability requirements were set to:

$$253 \quad \left(\frac{|f(n) - f(n+1)|}{f(n)} \right) < 0.01, \quad (5)$$

$$254 \quad \left(\frac{|\xi(n) - \xi(n+1)|}{\xi(n)} \right) < 0.05, \quad (6)$$

$$256 \quad [1 - MAC(\{\phi(n)\}, \{\phi(n+1)\})] < 0.02, \quad (7)$$

258 where (n) and $(n+1)$ are the n^{th} and $n^{th} + 1$ model order, f is the natural frequency, ξ is the
 259 damping, and ϕ is the mode shape vector.

Results and discussion

Processing the data yielded nine of the first ten modes of the freely suspended beams in the bandwidth between 0 Hz and 300 Hz , the only exclusion being the first flexural mode along the weak axis. Tab1 shows the results estimated from the EFDD method and SSI-cov, with the results of the preliminary numerical model. As can be seen from the table, the estimated frequencies are very close to each other. The mode shape estimates are very consistent, with CrossMAC values higher than 0.99. Furthermore, the experimental results do not differ too much from the numerical ones. The only exception being the swapping of position between the 1st flexural mode along the strong axis and the 2nd flexural mode along the weak axis in the measured modes, compared to the numerical ones. Fig5 shows the experimental modes: the MAC matrix in Fig6 remarks on the excellent correspondence between experimental and numerical modes. The fact that some off-diagonal terms have very high values could seem odd at first glance, but with a more careful look, one can notice how these are the modes that have similar shape along the two orthogonal axes.

Three modes were identified in the bandwidth between 0 Hz and 300 Hz when the two beams were simply supported. Tab2 presents the results of dynamic identification compared to the results of the numerical model and the first three frequency calculated according to Eq.(3). The excellent crossMACs between analytical and numerical mode shapes confirm that the beam's meshing size for the numerical model was appropriately chosen. The experimental mode shapes are depicted in Fig7. A more significant difference between measured and numerical/analytical results is appreciable for the II and the III mode, both in terms of natural frequencies and mode shapes. The differences are probably due to the stiffness of the metal supports, which are not able to restrain the uplift movement.

Interestingly, the measured mode shapes, depicted in Fig7, reveal the presence of defects on both beams, which were not detectable when the beam was tested as freely suspended. The visible variations recorded by the accelerometers nearby the location of the damages, especially in the III mode, suggest that higher modes can be used as indicators to localise the presence of damages on

287 structural elements, as already suggested by other authors (Ciambella et al. 2019; Aloisio et al.
288 2020e).

289 The identification of the plywood boards in Tab3 returned seven stable modes in the bandwidth
290 0 – 100 Hz. The numerical model evidenced the presence of some modes, not reported here, that
291 could not be identified from the chosen setup. These are those modes where all the positions of the
292 accelerometers correspond to the nodes of the mode shapes (i.e. a point of dynamic equilibrium),
293 and therefore could not be detected. Out of the seven modes, three show a notable agreement with
294 the numerical model, namely: mode I, mode VI and mode VII (see Fig8). The others seem to be
295 more affected by the presence of the Rockwool pad. Looking more carefully at the mode shapes in
296 Fig8 one can notice how in mode I, VI and VII, the central point is a node of the modal shape and
297 accordingly less affected by the presence of the Rockwool. Whereas modes IV and V, where the
298 centre is an anti-node, are more affected by the insulation piece. Nevertheless, the addition of a small
299 set of springs at the centre of the numerical model, so as to simulate the presence of the Rockwool,
300 determine mode IV and mode V to exhibit a satisfactory agreement with the experimental data, as
301 remarked in the following paragraphs.

302 The dynamic identification of the simply-supported floor assembly returned two stable modes
303 in the bandwidth 0 – 40 Hz, that is the suggested bandwidth of interest for timber floors (EN1995
304 2004). Mode I is a torsional mode where the two beams move out of phase with each other, while
305 mode II is the first bending mode, namely the two beams are in phase. Tab4 reports the estimated
306 frequencies and damping ratios with the results of the numerical model. The particular configura-
307 tion of the floor, with the board not rigidly fixed to the beams, prompted the numerical model to
308 exhibit several local modes of the boards that had almost no effect on the beams. The mode shapes
309 from the numerical model were extracted from the modal displacement of nodes belonging to the
310 frame elements, in order to be faithful to the test setup. The results of the two methods are in excel-
311 lent agreement, with CrossMAC values higher than 0.99. In a single instance, the damping ratio of
312 the II mode from SSIcov was noticeably higher than that estimated from the EFDD. The adoption
313 of standardized material properties in the numerical model causes a significant error in terms of

314 frequency, although the mode shapes show a satisfactory correspondence with the experimental.
315 Moreover, in Tab4 the first bending frequency (mode II) can also be compared to the first bending
316 frequency calculated according to the analytical Euler-Bernoulli model. The two frequency values
317 reported correspond to the situation when a complete composite action between the beams and the
318 decking and only the beams are respectively considered for the calculation of $(EI)_L$, in Eq.(4).

319
320 It is worthwhile to point out that exciting the tested structure did undoubtedly help to increase
321 the signal to noise ratio, but it also partially masked the presence of spurious harmonics. Structures
322 under test may show dominant frequency components which do not represent natural frequencies but
323 derive from deterministic signals superimposed to the stochastic response (e.g., rotating equipment).
324 One of the criteria to identify the presence of such spurious harmonics is by looking at the plot of
325 the singular values of the PSD matrix. The PSD matrix presents a high rank in similar instances,
326 and the spurious frequency is recognizable in the plot of the singular values, which have a sharp-
327 pointed resonance peak. During the excitation, the peaks in the plot of the singular values could
328 be misunderstood for natural frequencies. In the current case, a few tests carried out without the
329 manual excitation revealed the occurrence of the spurious harmonics. Fig9 demonstrates this aspect
330 by comparing the plots of the singular values of the floor assembly.

331 **SENSITIVITY ANALYSIS AND MODEL UPDATING**

332 **Sensitivity analysis**

333 The solid element models of the beams were the base of a variance-based sensitivity analysis.
334 The analysis allowed decomposing the variance of the output (objective function, and natural fre-
335 quencies) of the model into fractions which can be attributed to the inputs (mechanical properties).
336 The first step was setting the inputs sampling range (mean value $\pm 30\%$) and generate the model
337 inputs according to the Saltelli's sampling scheme (Saisana et al. 2005) ($N * (2D + 2)$ model
338 inputs were generated, where $N = 100$ is the number of samples, and $D = 9$ is the number of
339 input parameters). After running all the model inputs the first-order (S1) and total-order (ST)
340 sensitivity indices were calculated. S1 and ST measure respectively, the effect of varying a single

341 parameter alone and the contribution to the output variance of the selected parameter including
342 all variance caused by its interactions with the other parameters. Since the results were similar
343 for both beams, Tab5 and Tab6 details those of a single beam. The first two columns express the
344 impact of the mechanical parameters on the total response (Obj. Fun. = Objective Function). The
345 following columns show the impact of the parameters on each mode (SA=Strong axis, WA=Weak
346 axis, Tors=Torsional mode).

347 From Tab5 and Tab6 it is evident that the dynamic behaviour is mainly influenced by E_X and
348 G_{XZ} , while G_{XY} shows a moderate contribution. The other parameters do not affect the results at
349 all. For the objective function the differences in the first and total order indexes show some degree
350 of interaction between E_X and G_{XZ} . Furthermore, between all the flexural modes, E_X is the most
351 critical parameter. However, in the dynamic parallel to the strong axis, the shear modulus G_{XZ} gain
352 importance in higher modes (see SAIII in Tab5). The fact that G_{XY} show very little influence for
353 the modes along the weak axis agrees with the fact that the cross-section is much higher than wider
354 (115 x 315 mm). This aspect is also evident in the torsional modes, where G_{XZ} is the most crucial
355 parameter. These observations are in line with what one could expect from the slender nature of
356 the element, which should indeed follow the assumptions of the beam theory.

357 The fact that some first-order indices add up to values slightly higher than one may derive from
358 the reduced number of samples ($N = 100$). Still, this does not affect the substantial interpretation
359 of the results. A 2^{nd} order polynomial was fitted to the values of the objective function to provide
360 a graphical description of the results in the E_X and G_{XZ} domain, see Fig10.

361 **Model Updating**

362 Finite element model updating methods aim at tuning a numerical model to the measured
363 response(Marwala 2010). It is assumed that the measurements are correct, and the model under
364 consideration will need to be updated to reflect the measured data better.

365 As already mentioned, two global optimization algorithms headed the model updating process:
366 particle swarm optimization (PSO) and differential evolution (DE). Eq.(1) was used in both to
367 minimize the distance between measurements and numerical simulations. The results of the

368 sensitivity analysis supported the adoption of β equal to 0.1. The choice counterbalanced the
369 significant contribution of the second part of the objective function (due to the MAC). The swapping
370 of position between the 1st flexural mode along the strong axis and the 2nd flexural mode along the
371 weak axis resulted, in fact, in very high values of the objective function, see Fig10.

372 The natural frequencies depend on the ratio between the stiffness and the mass of the system.
373 The direct weighting of the beams and the panel allowed a straightforward calibration of the FE
374 model inertia (Beam 1 = $455\text{kg}/\text{m}^3$, Beam 2 = $470\text{kg}/\text{m}^3$, panel = $680\text{kg}/\text{m}^3$). Tab7, Tab8 and
375 Tab9, Tab10 report the frequencies of the initial FE models (with the measured mass), with errors to
376 test results, referred to the frame and solid element models, respectively. The first update regarded
377 the frame element. Isotropic material properties are used for these elements by SAP2000 even if
378 the material is defined as orthotropic. However, the definition of the material as orthotropic allows
379 to separately define the elastic modulus E_X (axial stiffness and bending stiffness) and the shear
380 modulus G_{XZ} (transverse shear stiffness), which were the selected parameters to be updated in this
381 model. The last columns of Tab7 and Tab8 list the frequencies of the updated FE model, compared
382 to test results. The averages of the optimal solutions of the two algorithms, used to calculate the
383 modes of the updated model, are presented in the lower part of Tab7 and Tab8. The tables reveal
384 that the updating process did improve the agreement between the physical and numerical model.
385 However, the model did not resolve the already mentioned inconsistency due to the swapping of
386 position between modes. Furthermore, the updating of Beam 1 showed that there is a reduction
387 of the elastic modulus E_X compared to the mean value of the standards, while that of Beam 2 E_X
388 increases slightly. Likely, the reduction of the elastic modulus E_X in Beam 1 derives from the wide
389 crack present by the end of the beam. The shear modulus G_{XZ} is higher than expected in both
390 beams, more evident in Beam 2 than Beam 1.

391 In the second step, the updating regarded the solid beam models. Following the results of the
392 sensitivity analysis, only E_X , G_{XZ} and G_{XY} were updated among the nine mechanical properties.
393 The updating process involved G_{XY} , although the sensitivity analysis showed that this parameter
394 has minimal effect on the dynamic behaviour in the selected frequency range. Similarly to the frame

395 element model, Tab9 and Tab10 reports the results of the solid beam models. The last columns
396 show the frequencies and the error of the updated model, while the lower part of the table reports
397 the averages of the optimal solutions found by the two algorithms. This model yielded a significant
398 improvement in the results. Still, as occurred in the frame-like models, the updating did not resolve
399 the inconsistency due to the swapping of position between modes. There is a similar reduction of
400 the elastic modulus E_X in Beam 1, probably caused by the cracks. Similar observations about the
401 frame element model are valid about the shear modulus G_{XZ} of both beams. The shear modulus
402 G_{XY} exhibits an increment to values suggested by the standards in the Beam 2. In contrast, there
403 is a decrease in the shear modulus G_{XY} in Beam 1. The results in terms of MAC are very high
404 (≈ 0.99), except for the inconsistency between the first modes.

405 The sensitivity analysis and the model updating process confirm that the "solid elements" model
406 does not determine a significant enhancement of the results to the "frame elements" model. For
407 these reasons, the use of "solid elements" for the FE model of the floor assembly is worthless, given
408 the enormous computational costs related to the use of the "solid elements" model.

409 The use of low-stiffness linear links ($100 N/mm$) placed by the middle of the plate, in cor-
410 respondence of the Rockwool pad, enhance the quality of the results referred to mode IV and
411 V. The first column of Tab3 and the second column of Tab11 prove this aspect. Conversely, the
412 low-stiffness linear links did not affect the results of mode I, VI and VII: the centre is a node in these
413 modes. Accordingly, the authors used only mode I, VI and VII to update the FE model with the
414 optimization algorithms as carried out in the beam models. The last columns of Tab11 summarize
415 the results, while the lower part of the table reports the optimal solutions (rounded).

416 The numerical model of the floor assembly was built after the updating of the single structural
417 components. As already mentioned, frame elements were used to model the beams and shell
418 elements the plywood boards. The boards were "lifted" to the centre of mass of the beams. Link
419 elements, with infinite stiffness in the axial direction and zero stiffness to all the others, model
420 the connection between the elements. The updated parameters of the single sub-assemblies yield
421 already a good match with the measurements (compare the first columns of Tab12). However, it was

422 decided to enhance it further, by changing the supports' stiffness from infinite into a finite value.
423 The simplicity of the problem encouraged a manual update based on trial and error. Tab12lists
424 the results of the updated FE model with the optimal solution. As further validation, the estimated
425 stiffness value of the supports was applied to the simply-supported beam models. The adoption
426 of a finite value of stiffness of the supports determine a further enhancement of the results, see
427 Tab13. It was observed that a higher stiffness for the supports was needed to reduce the frequency
428 discrepancy further.

429 The findings of the investigation confirm that the dynamic response of a timber floor is highly
430 sensitive to every parameter that describe its components and its boundary conditions. Unfortu-
431 nately predicting accurately the dynamical behaviour of a timber floor with simplified analytical
432 approach is rarely possible. Even if well-known and understood analytical models are certainly
433 useful at preliminary design stages, more detailed numerical models are needed if high level of
434 performance of the floor are desired. It is possible to obtain numerical models very faithful to
435 reality, however updating every element that composes the system is not feasible in practical ap-
436 plications. To assess the behaviour of an existing floor in a building, a researcher would need
437 update all the parameters "at once" with an inevitable loss of detail. A careful examination of the
438 drawings corroborated by on-site inspections is therefore of paramount importance in order to build
439 a detailed and representative numerical model. Furthermore the level of detail of the experimental
440 campaign will set the basis for the success of the updating process.

441 **CONCLUSIONS**

442 This paper investigates the dynamic behaviour of a simply-supported timber floor assembly and
443 its composing elements. A sensitivity analysis revealed the influence of mechanical parameters on
444 the dynamic response. As the last step, the numerical models were updated to reflect the findings
445 of the measurements better. The main findings are:

- 446 • OMA techniques can be used, instead of EMA techniques, to test not only massive civil
447 engineering structures, but also smaller structural elements, such as floors, beams etc., and

448 their results can be used to calibrate the parameters of numerical models.

- 449 • It is helpful to continuously and randomly excite the tested components, for example, by
450 rubbing something onto it, to increase the signal-to-noise ratio. Significant attention must,
451 however, be paid not to mistake spurious harmonics for natural frequencies.
- 452 • Small pieces/layers of insulation material, can be used to recreate free-free boundary con-
453 ditions if the suspension of the element is not possible.
- 454 • Higher modes were found more susceptible to damages and defect when the beams were
455 tested as simply-supported. They could therefore be used as damage indicators to assess
456 the state of health and/or to localise defects in it. When the beams were tested as freely
457 suspended, however, the damages seemed not to affect the modal shapes.
- 458 • The results of the identification, for any component, are very susceptible to the nature of
459 the boundary conditions and even small variations in them significantly affect the results.
- 460 • The results confirm that the use of the well-known beam model is more than capable of
461 correctly predicting the behaviour of slender components. The significant computational
462 time needed for a solid element model is not worth the gain in terms of precision.

463 This research was preliminary to more-in-depth investigations about the walked-induced vibra-
464 tion response of timber floors. The authors aim at using the assembled floor system and the updated
465 numerical model to study different walking models further and compare numerical simulations
466 with walking tests. This investigation will allow studying the various metrics used by building
467 codes and relevant standards to evaluate and assess building floor vibrations.

468 **DATA AVAILABILITY STATEMENT**

469 Some or all data, models, or code that support the findings of this study are available from the
470 corresponding author upon reasonable request.

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TABLE 1. Results dynamic identification freely suspended beams

Mode	SAP2000	Beam 1 SSIcov		Beam 1 EFDD		Beam 2 SSIcov		Beam 2 EFDD	
	f_n [Hz]	f_n [Hz]	ξ [%]	f_n [Hz]	ξ [%]	f_n [Hz]	ξ [%]	f_n [Hz]	ξ [%]
1-Flex-WA	25.81	-	-	-	-	-	-	-	-
2-Flex-WA	69.96	62.98	0.72%	63.06	0.59%	65.66	0.61%	65.83	0.60%
1-Flex-SA	67.57	65.50	1.03%	65.98	1.00%	67.77	0.89%	67.54	1.13%
1-Tors	72.78	70.32	1.05%	70.52	1.07%	71.94	1.28%	71.51	0.85%
3-Flex-WA	133.75	124.09	0.99%	124.87	1.01%	129.50	0.89%	129.42	0.92%
2-Tors	146.86	149.15	2.12%	148.99	1.61%	154.37	2.24%	153.44	1.96%
2-Flex-SA	166.79	154.07	0.59%	155.97	0.54%	157.90	0.70%	157.91	0.66%
4-Flex-WA	214.06	194.83	0.59%	195.13	0.53%	202.75	0.65%	202.03	0.65%
3-Tors	223.42	220.05	2.53%	219.88	1.60%	227.53	3.55%	222.68	1.49%
3-Flex-SA	286.77	273.90	0.80%	273.68	0.73%	279.31	0.83%	278.40	0.73%

Flex = Flexural mode; Tors = Torsional mode; WA = Weak Axis; SA = Strong Axis
Numerical and analytical model's parameters:
 $E_x = 13000$ [MPa], $G_{xz} = G_{xy} = 650$ [MPa]; $\rho = 430$ [kg/m³]

TABLE 2. Results dynamic identification simply supported beams

Mode	SAP2000 f_n [Hz]	Analytical f_n [Hz]	Beam 1 - SSIcov f_n [Hz]	ξ [%]	Beam2 - SSIcov f_n [Hz]	ξ [%]
I	32.72	31.42	28.90	0.70%	29.93	0.75%
II	117.77	125.66	90.22	1.76%	92.53	2.04%
III	230.84	282.74	149.50	1.75%	153.73	1.80%

Numerical and analytical model's parameters:
 $E_x = 13000$ [MPa]; $G_{xz} = G_{xy} = 650$ [MPa]; $\rho = 430$ [kg/m³];
 $K_{support} = \infty$

TABLE 3. Results dynamic identification plywood boards

Mode	SAP2000	Plate - SSIcov			Plate - EFDD		
	f_n [Hz]	f_n [Hz]	ξ [%]	MAC	f_n [Hz]	ξ [%]	MAC
I	28.04	30.42	0.88%	95.8%	30.50	0.88%	96.0%
II	35.48	37.45	1.39%	51.2%	37.32	1.46%	51.6%
III	39.20	39.96	1.62%	47.8%	40.09	1.21%	48.8%
IV	35.36	46.90	2.80%	30.3%	47.05	1.63%	30.1%
V	61.90	74.29	5.86%	92.9%	74.45	5.24%	91.4%
VI	81.16	88.83	1.40%	98.6%	89.03	0.76%	98.9%
VII	93.56	91.24	1.58%	98.6%	91.08	0.74%	98.8%

Numerical model's parameters:
 $E_x = 6000$ [MPa], $E_y = 8000$ [MPa]; $\rho = 650$ [kg/m³]

TABLE 4. Results dynamic identification floor

Mode	SAP2000	Analytical	Floor - SSICov			Floor - EFDD		
	f_n [Hz]	f_n [Hz]	f_n [Hz]	ξ [%]	MAC	f_n [Hz]	ξ [%]	MAC
I	20.09	-	16.76	3.26%	90.9%	16.81	3.28%	92.2%
II	23.44	26.37 / 23.41	20.28	3.96%	91.5%	20.31	2.40%	93.5%

Numerical model's parameters:
 Glulam: $E_x = 13000$ [MPa], $G_{xz} = 650$ [MPa]; $\rho = 430$ [kg/m³]
 Plywood: $E_x = 6000$ [MPa], $E_y = 8000$ [MPa]; $\rho = 650$ [kg/m³]
 Supports: $K_{support} = \infty$

TABLE 5. First-order (S1) and total-order (ST) sensitivity indices for beam 1. Part I

Param.	Obj. Fun.		SA I		SA II		SA III		WA II	
	S1	ST	S1	ST	S1	ST	S1	ST	S1	ST
E_X	60%	80%	97%	97%	85%	86%	62%	64%	97%	97%
E_Y	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
E_Z	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
ν_{YX}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
ν_{ZX}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
ν_{YZ}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
G_{XY}	0%	5%	0%	0%	0%	0%	0%	0%	0%	0%
G_{XZ}	32%	56%	1%	1%	10%	11%	32%	35%	0%	0%
G_{YZ}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%

TABLE 6. First-order (S1) and total-order (ST) sensitivity indices for beam 1. Part II

Param.	WA III		WA IV		Tor I		Tor II		Tor III	
	S1	ST	S1	ST	S1	ST	S1	ST	S1	ST
E_X	96%	96%	94%	94%	0%	0%	0%	0%	0%	1%
E_Y	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
E_Z	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
ν_{YX}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
ν_{ZX}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
ν_{YZ}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%
G_{XY}	1%	1%	3%	3%	3%	3%	4%	4%	5%	5%
G_{XZ}	0%	0%	0%	0%	101%	101%	100%	100%	98%	98%
G_{YZ}	0%	0%	0%	0%	0%	0%	0%	0%	0%	0%

TABLE 7. Result of the model updating on the "Frame elements" beam model. Beam 1

Mode	Experimental f_n [Hz]	Beam 1		Optimised model	
		FE Initial model f_n [Hz]	Error	f_n [Hz]	Error
1-Flex-SA	65.50	66.14	-0.97%	62.94	3.92%
2-Flex-WA	62.98	68.08	-8.09%	64.61	-2.59%
3-Flex-WA	124.09	129.17	-4.09%	123.76	0.26%
2-Flex-SA	154.07	163.31	-5.99%	157.12	-1.98%
4-Flex-WA	194.83	207.78	-6.65%	198.49	-1.88%
3-Flex-SA	273.90	280.09	-2.26%	272.34	0.57%

Optimal parameters:
 $E_x = 11800$ [MPa]; $G_{xz} = 670$ [MPa]; $\rho = 455$ [kg/m³]

TABLE 8. Result of the model updating on the "Frame elements" beam model. Beam 2

Mode	Experimental f_n [Hz]	Beam 2		Optimised model	
		FE Initial model f_n [Hz]	Error	f_n [Hz]	Error
1-Flex-SA	67.77	65.21	3.77%	65.01	4.07%
2-Flex-WA	66.24	67.12	-1.33%	66.74	-0.76%
3-Flex-WA	129.86	128.19	1.28%	127.83	1.56%
2-Flex-SA	157.90	161.02	-1.98%	162.40	-2.85%
4-Flex-WA	202.75	204.87	-1.05%	205.07	-1.15%
3-Flex-SA	279.31	276.16	1.13%	281.68	-0.85%

Optimal parameters:
 $E_x = 13100$ [MPa]; $G_{xz} = 700$ [MPa]; $\rho = 467$ [kg/m³]

TABLE 9. Results of the updating on the "Solid elements" beam model. Beam 1

Mode	Experimental f_n [Hz]	Beam 1		Optimised model	
		FE Initial model f_n [Hz]	Error	f_n [Hz]	Error
1-Flex-SA	65.50	65.76	-0.39%	62.54	4.52%
2-Flex-WA	62.50	68.08	-8.94%	64.35	-2.97%
1-Tors	70.32	70.83	-0.73%	72.15	-2.60%
3-Flex-WA	124.17	130.17	-4.83%	123.12	0.85%
2-Tors	149.15	142.92	4.17%	145.20	2.65%
2-Flex-SA	154.07	162.32	-5.35%	156.52	-1.59%
4-Flex-WA	193.55	208.32	-7.63%	197.20	-1.89%
3-Tors	220.05	217.43	1.19%	219.99	0.03%
3-Flex-SA	273.90	279.08	-1.89%	272.75	0.42%

Optimal parameters:
 $E_x = 11600$ [MPa]; $G_{xz} = 690$ [MPa]; $G_{xy} = 620$ [MPa];
 $\rho = 455$ [kg/m³]

TABLE 10. Results of the updating on the "Solid elements" beam model. Beam 2

Mode	Experimental f_n [Hz]	Beam 2			
		FE Initial model f_n [Hz]	Error	Optimised model f_n [Hz]	Error
1-Flex-SA	67.77	64.84	4.32%	64.71	4.50%
2-Flex-WA	65.66	67.13	-2.24%	66.74	-1.65%
1-Tors	71.94	69.84	2.92%	74.15	-3.07%
3-Flex-WA	129.50	128.35	0.89%	127.86	1.26%
2-Tors	154.37	140.92	8.71%	149.27	3.30%
2-Flex-SA	157.90	160.05	-1.36%	161.62	-2.36%
4-Flex-WA	201.36	205.40	-2.01%	205.14	-1.87%
3-Tors	227.53	214.38	5.78%	226.28	0.55%
3-Flex-SA	279.31	275.17	1.48%	281.06	-0.63%

Optimal parameters:
 $E_x = 12800$ [MPa]; $G_{xz} = 740$ [MPa]; $G_{xy} = 700$ [MPa];
 $\rho = 467$ [kg/m³]

TABLE 11. Results of the updating of the Plywood board

Mode	Experimental	FE Initial model		Optimised model		MAC
	f_n [Hz]	f_n [Hz]	Error	f_n [Hz]	Error	
I	30.42	28.31	6.9%	29.51	3.0%	99.8%
II	37.45	35.57	5.0%	37.60	-0.4%	47.8%
III	39.96	39.27	1.7%	38.87	2.7%	51.5%
IV	46.90	44.07	6.0%	44.66	4.8%	95.9%
V	74.29	71.02	4.4%	71.58	3.7%	98.9%
VI	88.83	81.39	8.4%	89.23	-0.5%	98.2%
VII	91.24	93.75	-2.8%	92.10	-0.9%	98.3%

Optimal parameters:
 $E_X = 6500$ [MPa]; $E_Y = 7500$ [MPa]; $\rho = 680$ [kg/m³]

TABLE 12. Results of the updating of the floor

Mode	Experimental	FE Initial model		Optimised model		
	f_n [Hz]	f_n [Hz]	Error	f_n [Hz]	Error	MAC
I	16.76	18.62	-11.11%	17.15	-2.32%	94.0%
II	20.30	21.77	-7.22%	20.30	0.00%	93.5%

Optimal parameters:
Glulam: see Tab6
Plywood: see Tab8
Supports: $K_{support} = 6000 [N/mm]$

TABLE 13. Simply supported beam with updated parameters

Mode	Experimental		FE Updated		FE Updated	
	Beam 1	Beam 2	Beam 1	MAC	Beam 2	MAC
I	28.90	29.93	28.77	99.9%	29.74	100.0%
II	90.22	92.53	90.49	99.8%	92.17	99.5%
III	149.50	153.73	150.48	96.0%	151.67	92.5%

Optimal parameters:
Glulam: see Tab6
Supports: $K_{support} = 9000 [N/mm]$

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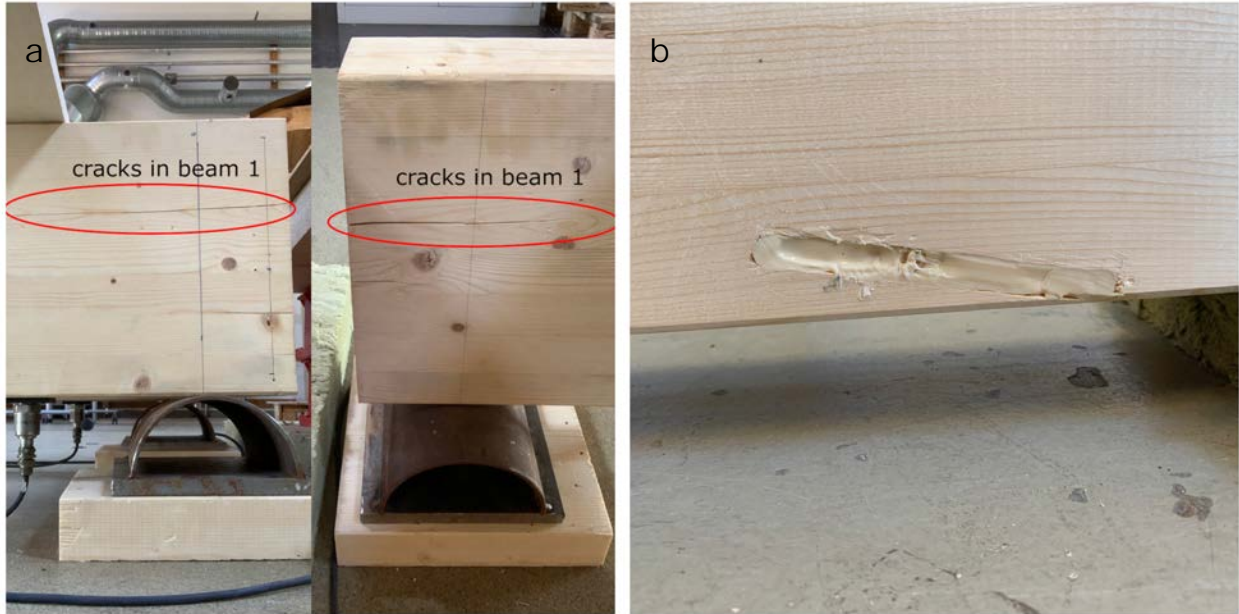
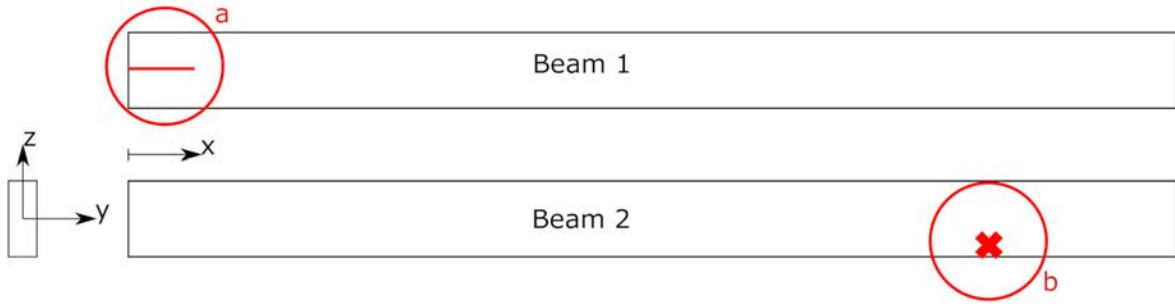


Fig. 1. Location of the defects on "Beam 1" and "Beam 2".



Fig. 2. Floor assembly: left view from above, right view from below.

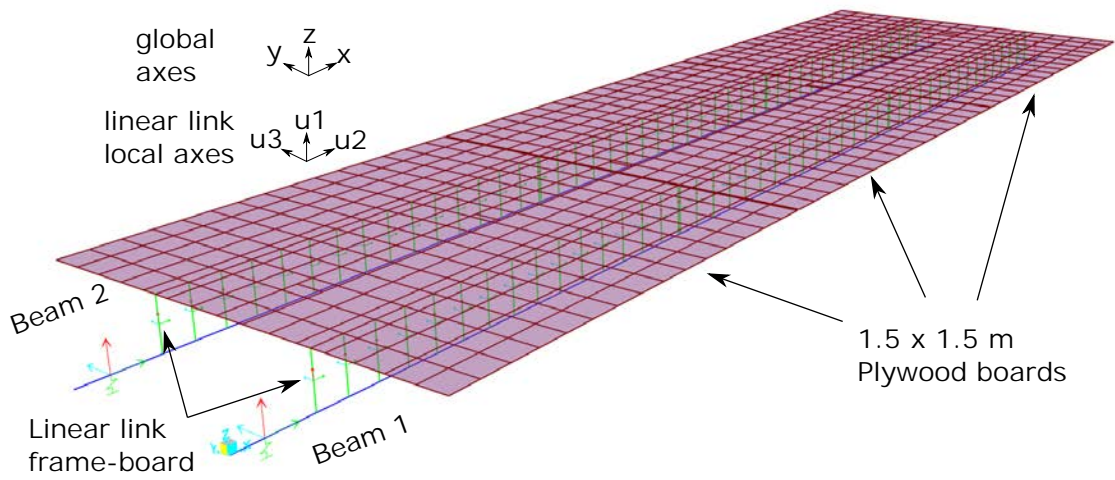


Fig. 3. Sap2000 FE model of the floor.

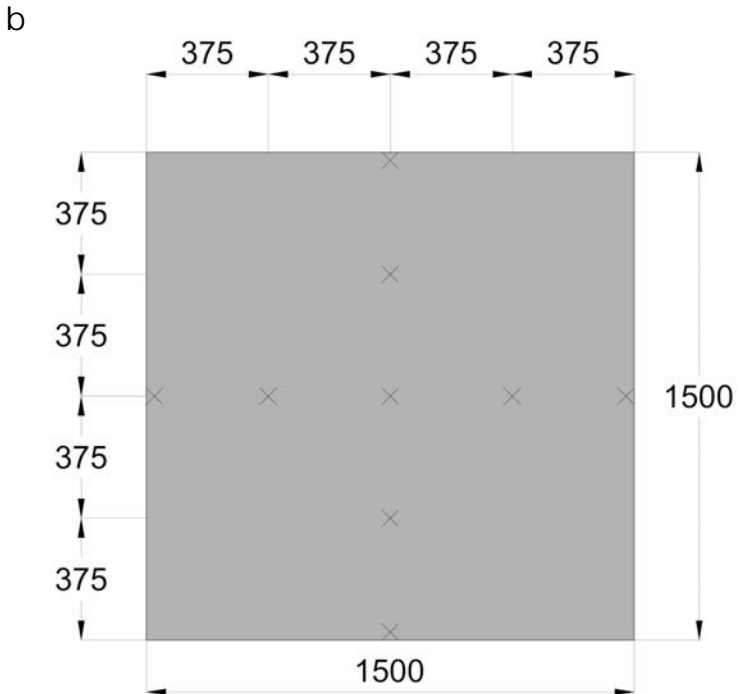
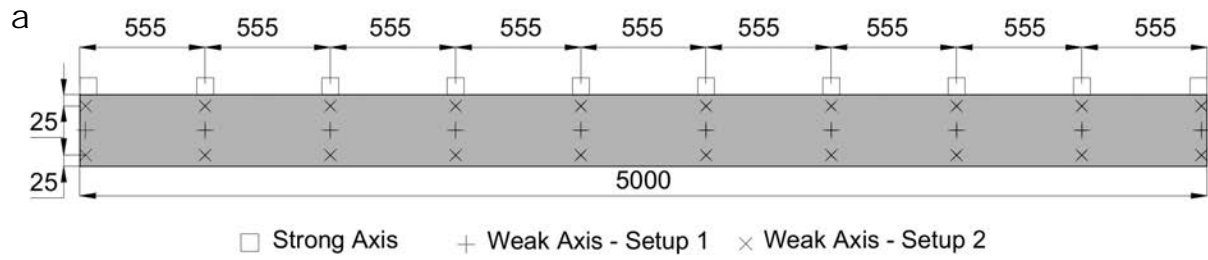


Fig. 4. a) Test setup glulam beams, b) Test setup plywood boards, c) Test setup for torsional modes glulam beams. (all the dimensions are in mm)

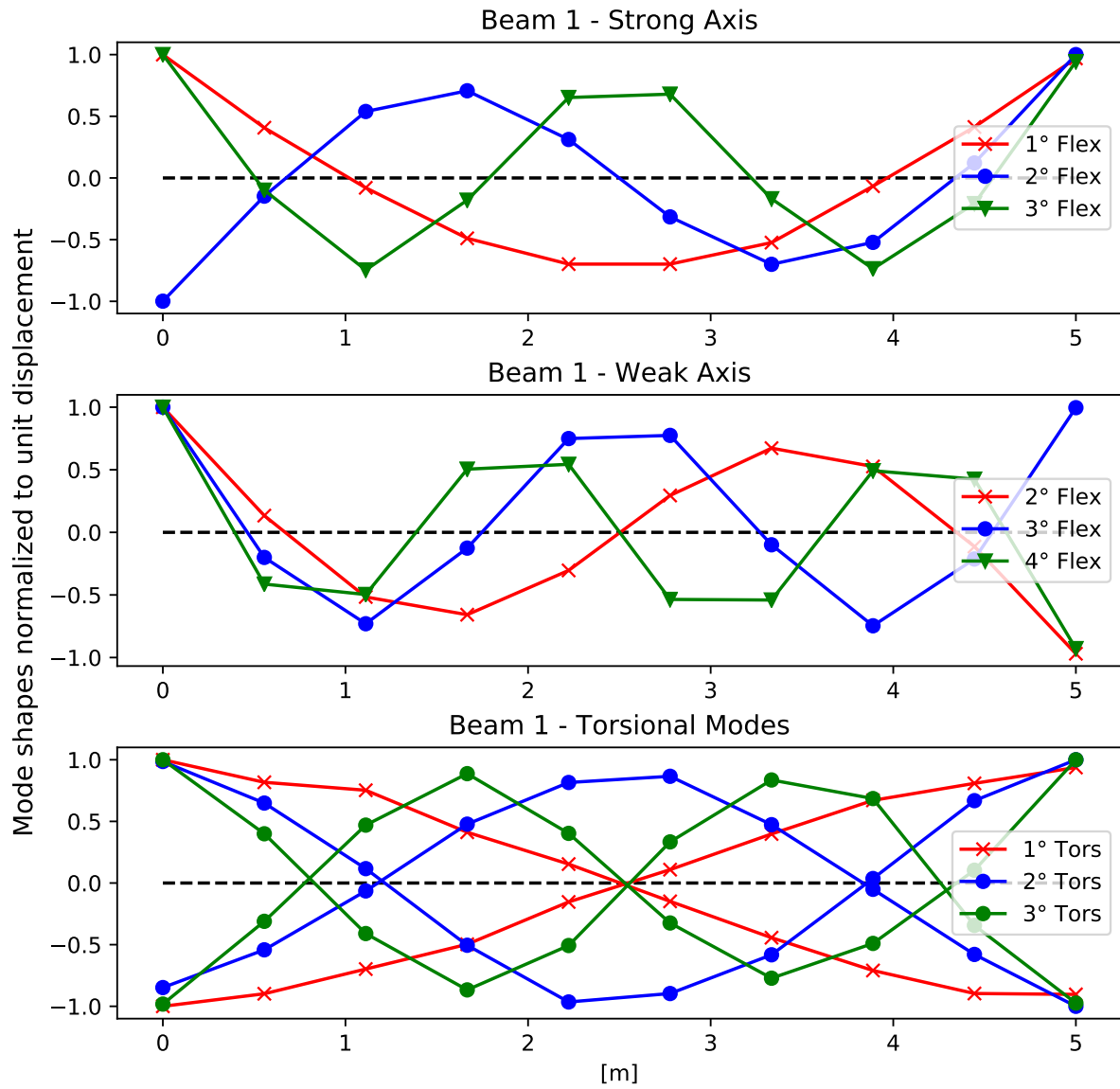


Fig. 5. Experimental modal shapes freely suspended beam.

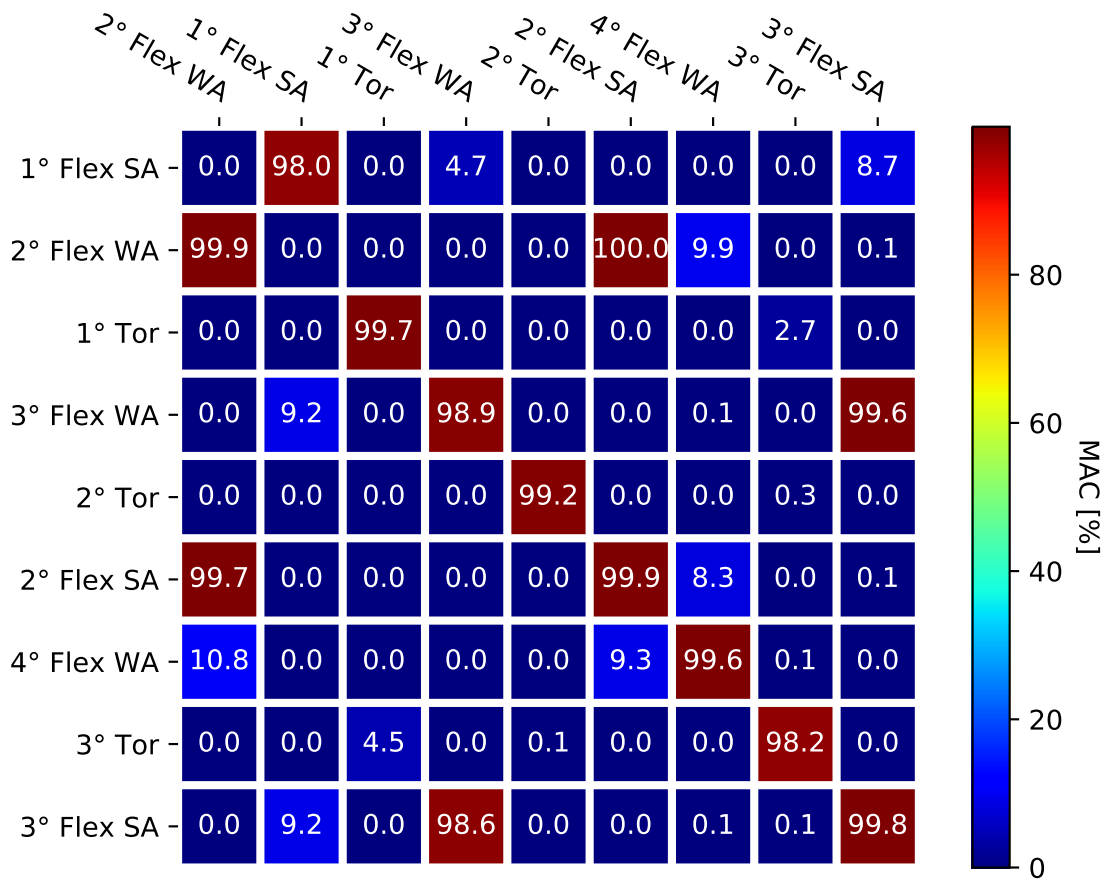


Fig. 6. MAC matrix: measured mode shapes vs numerical.

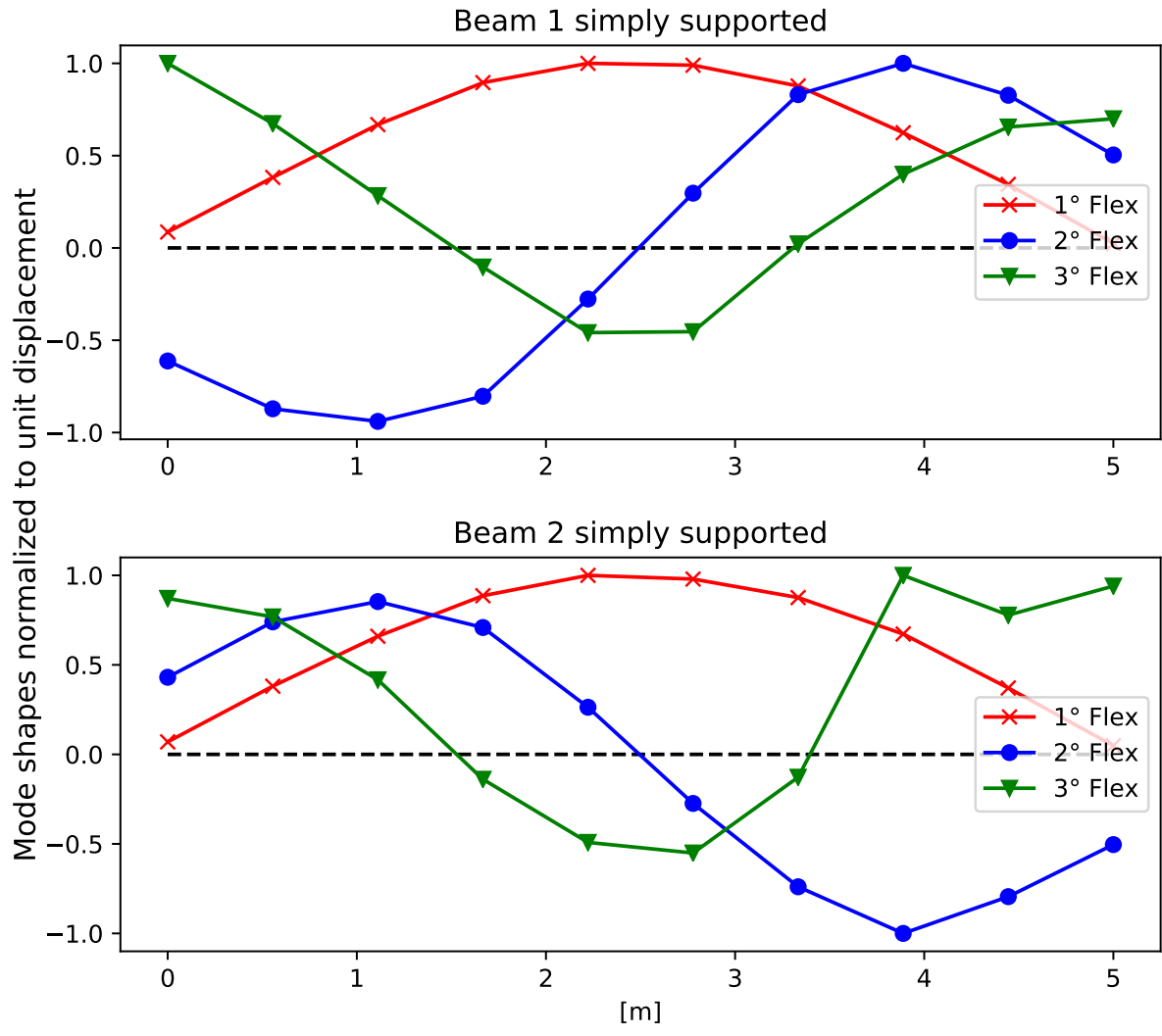


Fig. 7. Experimental modal shapes simply supported beam.

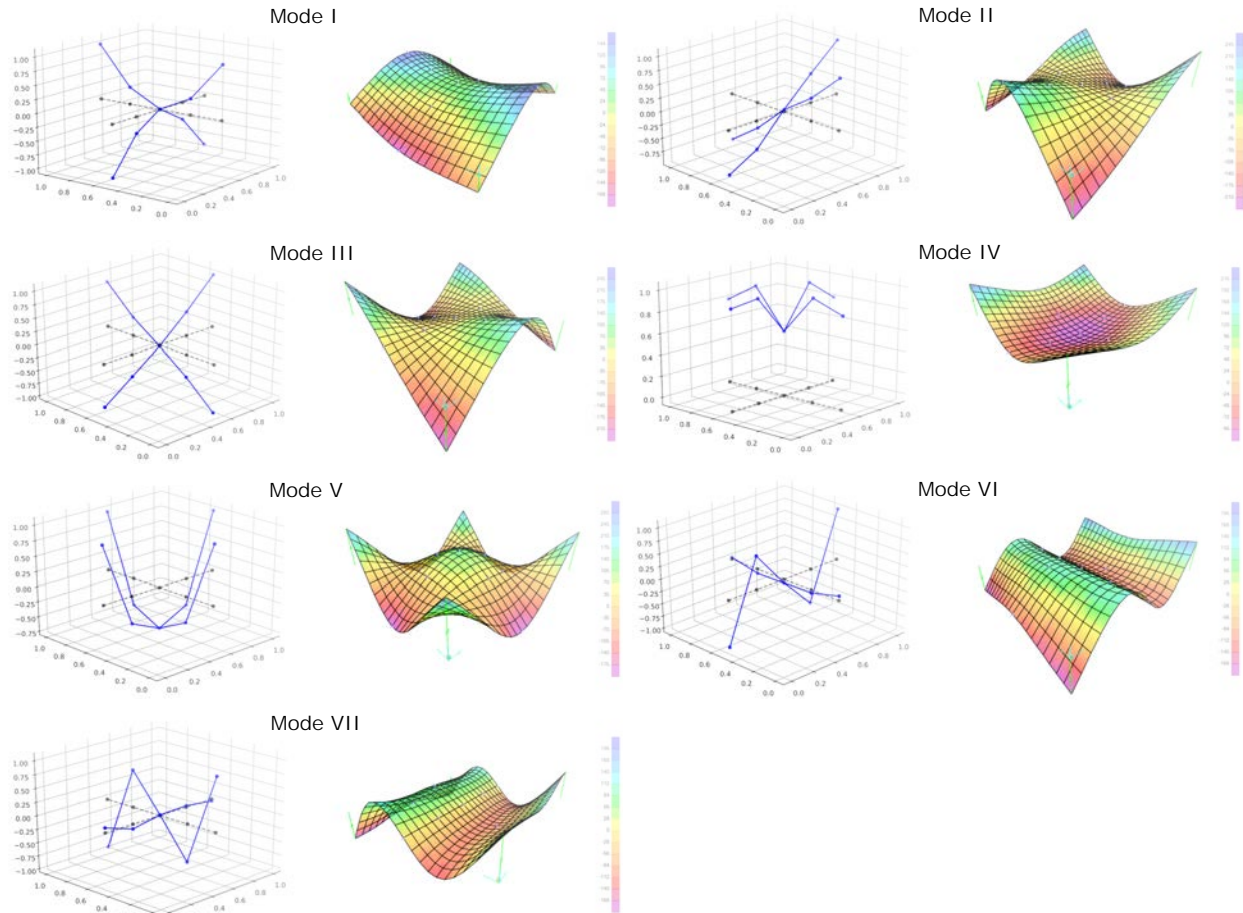


Fig. 8. Experimental and numerical mode shapes of the plywood board.

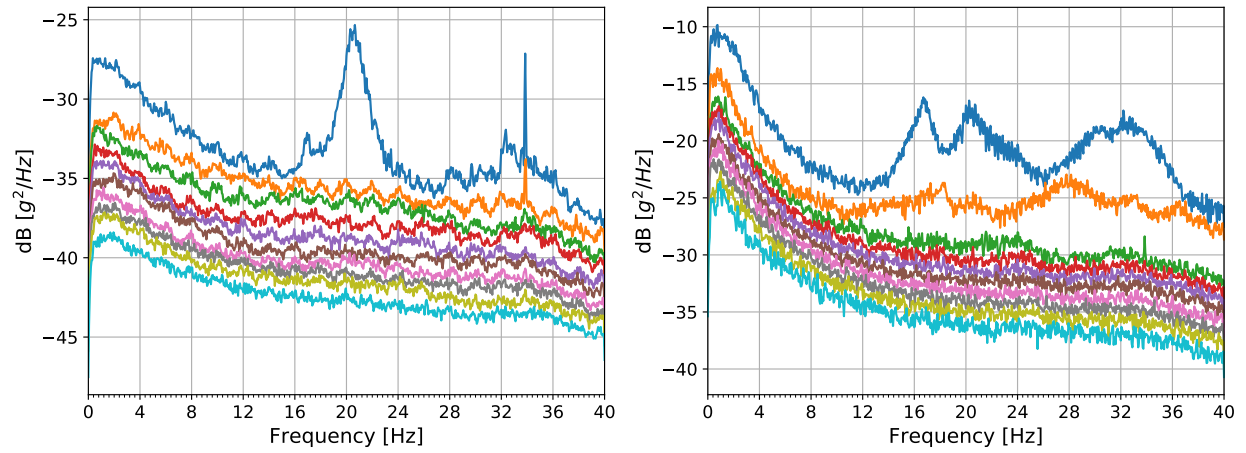


Fig. 9. Singular values plot: left unexcited floor, right excited floor.

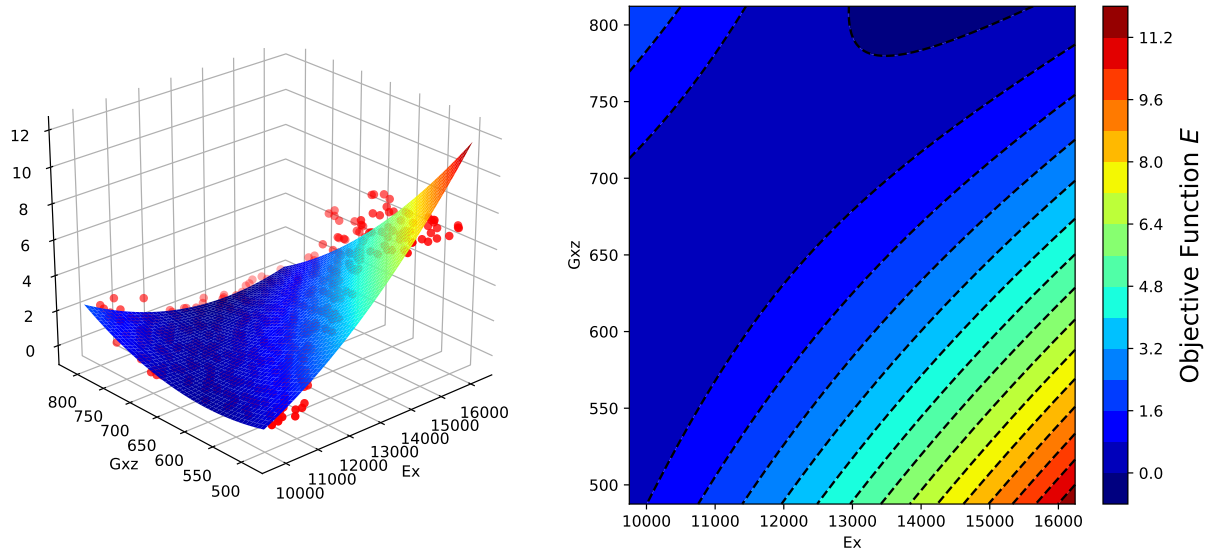


Fig. 10. 2^{nd} order polynomial fit: left 3D view with data points, right contour plot.