

# 1        **Strength and Stiffness of Cross-Laminated Timber (CLT) shear walls:**

## 2                                **State-of-the-art of analytical approaches**

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### 6        **ABSTRACT**

7        In the last years, the timber construction practice has been revived. Cross-laminated timber  
8        (CLT) plays a key role in this timber renaissance. CLT constructions has seen a noticeable  
9        increase in the last decade, especially in Europe, as it enables tall wooden buildings using a  
10        sustainable material. Unfortunately, a consequence of the rapid advancements of timber  
11        technologies and construction techniques of the past years is that modern timber engineering  
12        codes are struggling to keep up to date. Furthermore, the results of scientific research in this  
13        field is often inhomogeneous and fragmented, and do not help in proving that these new  
14        methods and construction techniques are reliable and safe to use.

15        To overcome this gap, COST Action FP1402 was created which main purpose is to create new,  
16        and improve on existing, knowledge of timber design and construction. This paper provides a  
17        summary of multiple fundamental aspects of design of CLT shear walls through a review of  
18        relevant scientific papers. This paper thus aims to be a “state-of-the-art” of available methods  
19        used to assess the load-carrying capacity and the displacement of CLT shear walls.

### 20        **Keywords**

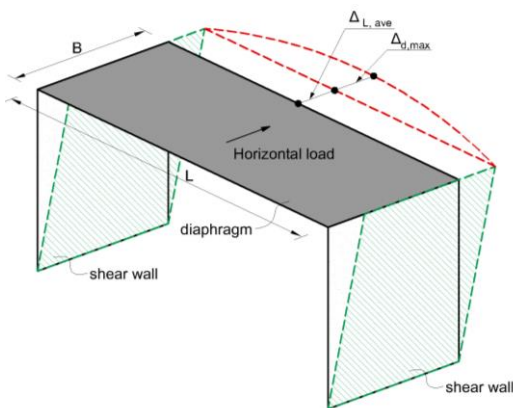
21        Cross-laminated timber (CLT); CLT shear walls; analytical approaches; strength assessment  
22        methods; stiffness assessment methods; state-of-the-art.

### 23        **1. INTRODUCTION**

24        Cross-laminated timber (CLT) is an efficient wood product that is well suited for multi-story  
25        timber buildings due to its relative high strength and stiffness. Knowledge of CLT technologies

26 and construction techniques has advanced quickly in the last few years but an absence of up-  
 27 to-date CLT standards makes it difficult for engineers to design cost-efficient CLT  
 28 constructions as design information is often limited to European Technical Approvals (ETA).  
 29 To harmonize recent efforts in research and to consolidate a correct building practice, the  
 30 COST Action FP1402 was established. FP1402 aims at deriving universal product parameters  
 31 and design methods to verify the compliance of timber systems with requirements in terms of  
 32 resistance, stability and serviceability asked for by designers, industry practitioners and  
 33 authorities.

34 While equations for design of light timber frame shear walls and diaphragms are available in  
 35 most codes or commentaries, no or little guidance on the in-plane stiffness of CLT diaphragms  
 36 is given, e.g., the Eurocodes [1, 2] provide little information on the design of the lateral load-  
 37 carrying system of CLT buildings [3]. Consequently, there is a need to explore design methods  
 38 for CLT shear walls and floor diaphragms, which constitute the main structural elements in tall  
 39 timber buildings [4, 5]. Floor diaphragms are typically considered as either fully flexible or  
 40 rigid, depending on the relation between the maximum in-plane deformation of the floor  
 41 diaphragm ( $\Delta_{d,max}$ ) and the average inter-story drift ( $\Delta_{L,ave}$ ) (Fig. 1). CLT diaphragms are often  
 42 considered as rigid in relation to the stiffness of shear walls, see e.g. [6, 7, 8,], even though  
 43 there is little information on its in-plane behavior [9].



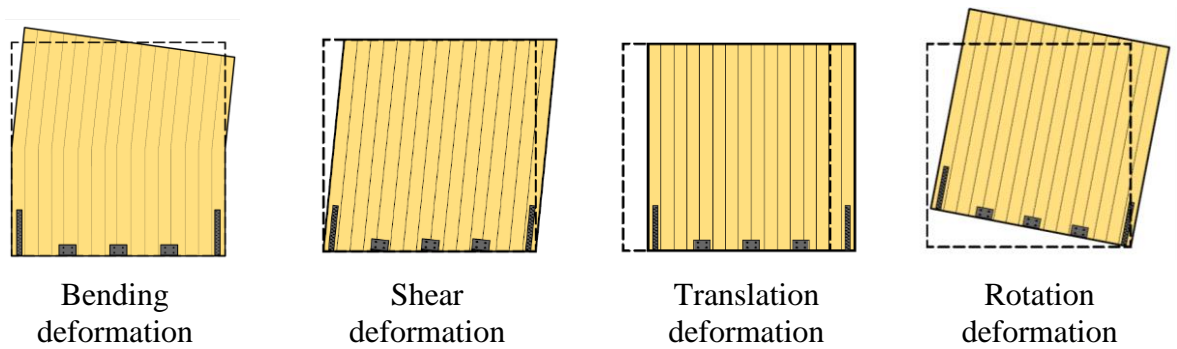
Diaphragm type	EUROPE	USA
	EN 1998:2010 [2]	ASCE 7-10 [10] IBC 2012 [11] SDPWS 2008 [12]
Flexible	$\Delta_{d,max} \geq 1.1\Delta_{L,ave}$	$\Delta_{d,max} \geq 2\Delta_{L,ave}$
Rigid	$\Delta_{d,max} < 1.1\Delta_{L,ave}$	$\Delta_{d,max} \geq 0.5\Delta_{L,ave}$
Semi-rigid	–	$0.5 < \frac{\Delta_{d,max}}{\Delta_{L,ave}} < 2$

44 **Figure 1.** Diaphragm definition based on displacement of diaphragm versus inter-story drift,  
 45 based on Moroder [3].

46 A general approach is to design for the case of either a rigid or a flexible diaphragm that gives  
47 the largest forces in shear walls. If the force in any shear wall differs by more than 15 % due  
48 to the change in the flexible and rigid diaphragm assumptions, then an envelope force approach  
49 should be used [13] where the design forces are based on the highest forces obtained from  
50 either the rigid or flexible case. However, neither assumption provides an accurate estimate of  
51 the lateral load distribution in case of semi-rigid diaphragms [9] which might lead to an  
52 underestimation of design forces since diaphragms are generally semi-rigid. [14]. By assigning  
53 the CLT diaphragm as either rigid or semi-rigid, lateral loads are distributed throughout  
54 diaphragms and shear walls in relation to the stiffness properties of each shear wall [14].  
55 In the literature, there is currently a lack of a cohesive view on how to properly design CLT  
56 shear walls. As part of the research of COST Action FP1402, this paper summarizes multiple  
57 fundamental aspects of design of CLT shear walls through a review of several relevant  
58 scientific papers. This paper thus aims to be a “state-of-the-art” of methods used to assess the  
59 load-carrying capacity and the displacement of CLT shear walls.

## 60 **2. STRENGTH OF CLT SHEAR WALLS**

61 Design of CLT shear walls is performed by assessing its load-carrying capacity and its stiffness.  
62 Analytical methods for design of CLT shear walls are based on the different contributions of  
63 the shear wall deformation. Four contributions are typically considered (Fig. 2); translational  
64 (or slip), rotational (or rocking), panel shear and panel bending. For most shear wall  
65 configurations, the contribution of the in-plane panel shear and bending deformation are much  
66 smaller than the deformations from translation and rocking, which is governed by steel  
67 connections that typically exhibit a much softer behavior [15, 16, 17]. Verification of load-  
68 carrying capacity and stiffness of CLT shear walls mainly consists of equilibrium equations  
69 based on wall geometry, external loading and connection properties.

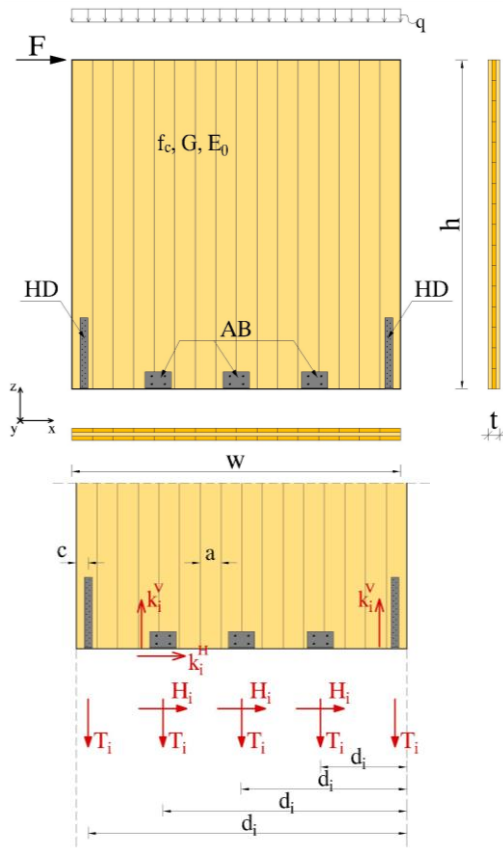


70 .

71 **Figure 2.** Illustration of the different contributions of the shear wall deformation.

72 **2.1. Definitions & notations**

73 To support the description of the different methods for strength and stiffness assessment, and  
 74 to assist in comparing their inputs and results, a standardized shear wall is defined with  
 75 generalized notations (Fig. 3). The shear wall is a CLT panel with width ( $w$ ), height ( $h$ ) and  
 76 thickness ( $t$ ) that is loaded by a lateral force ( $F$ ) and a vertical load ( $q$ ). The compressive  
 77 strength ( $f_c$ ) of the CLT panel and its elastic ( $E$ ) and shear ( $G$ ) modulus, along with the angle  
 78 brackets ( $AB$ ) and hold-downs ( $HD$ ) are defined in Fig. 14. The angle brackets and hold-downs  
 79 are described by their vertical strength ( $T$ ), horizontal strength ( $H$ ) and their stiffness in the  
 80 vertical ( $k^V$ ) and horizontal ( $k^H$ ) directions.



- $AB$  – angle bracket, steel connector
- $HD$  – hold-down, steel connector
- $n_{AB}$  – number of angle brackets
- $h$  – panel height [m]
- $w$  – panel width [m]
- $t$  – panel thickness [m]
- $F$  – horizontal force [N], i.e., shear wall capacity
- $q$  – uniformly distributed load [N/m]
- $c$  – edge distance [m]
- $d_i$  – distance from edge of panel to  $i^{\text{th}}$  connector [m]
- $a$  – width of the timber board [m]
- $T_i$  – tensile strength of the  $i^{\text{th}}$  connector [N]
- $H_i$  – horizontal strength of the  $i^{\text{th}}$  connector [N]
- $k_i^v$  – vertical stiffness of the  $i^{\text{th}}$  connector [N/m]
- $k_i^h$  – horizontal stiffness of the  $i^{\text{th}}$  connector [N/m]
- $f_c$  – compressive strength of timber [MPa]
- $G$  – Shear modulus perpendicular to the grain [MPa]
- $E_0$  – Elastic modulus parallel to the grain [MPa]

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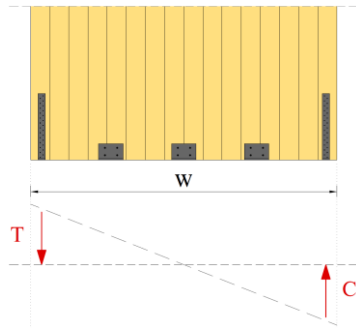
82 **Figure 3.** Definition of a standardized shear wall with notations used throughout this paper.

83 **2.2. Methods for strength assessment**

84 In the literature, several methods for calculating the load-carrying capacity of a CLT shear wall  
 85 are identified. Common for these methods is that they are mainly based on static equilibrium  
 86 equations and that the majority of methods consider the wall panel as rigid, i.e., the deformation  
 87 of the CLT panel itself is disregarded in favor of the connections. If not explicitly stated, all of  
 88 the methods resist overturning by hold-downs (HD), and translation by angle brackets (AB)  
 89 exclusively as was first proposed by Ceccotti et al. [6]. This means that an interaction of  
 90 vertical and horizontal forces in the connections are not typically considered as there is limited  
 91 experimental data and no current design guidance (Reynolds et al. [18]). Thus the load-carrying  
 92 capacity ( $F$ ) of the CLT shear wall can be simplified as  $F = \min(F_R; F_T)$  where  $F_R$  and  $F_T$   
 93 denotes the load-carrying capacity by rotation and translation respectively. If not otherwise  
 94 stated  $F_T = H_i \cdot n$ , where  $n$  is the number of angle brackets not used to resist shear wall rotation.

95 **Method A – Casagrande et al. [19]**

96 Casagrande et al. [19] presented a simplified analytical method to evaluate the load-carrying  
 97 capacity of a CLT shear wall based on rigid body rotation and static equilibrium between  
 98 internal forces and the overturning moment ( $F_R \cdot h$ ) (Fig. 4). With the point of rotation assumed  
 99 at the panel edge, the force in the hold-down ( $T$ ) due to a lateral ( $F_R$ ) and vertical load ( $q$ ) is  
 100 calculated as:



$$T = \frac{F_R \cdot h}{\tau \cdot w} - \frac{q \cdot w}{2} \quad (1)$$

101 **Figure 4.** The simplified analytical method as proposed in [19].

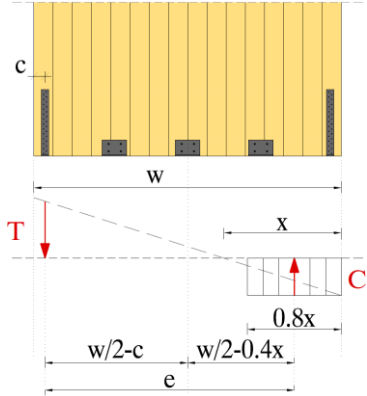
102 A lever arm coefficient,  $\tau$  of 0.90 times the length of the wall was used by Casagrande et al.  
 103 [19] to represent a reduction in width that takes into account the distance from the panel edge  
 104 to the hold-down, giving the expression for the maximum lateral force ( $F_R$ ) based on the  
 105 vertical capacity of the hold-down ( $T$ ):

$$F_R \cdot h = \left( T + \left( \frac{q \cdot w}{2} \right) \right) \cdot (0.9 \cdot w) \quad (2)$$

107 **Method B – Tomasi [20]**

108 Tomasi [20] proposed a “Stress block” method, where the nonlinear stress distribution for  
 109 wood in the compression zone is substituted by a rectangular stress block (Fig. 5). The  
 110 unknown position of the neutral axis is denoted by  $x$  and Tomasi [20] defined the size of the  
 111 “stress block” as  $0.8 \cdot x$  from which a resultant compression force ( $C$ ) is calculated (Eq. 3) based  
 112 on the compression resistance parallel to the grain ( $f_c$ ) and the width of the vertical lamellas  
 113 ( $t_{ef}$ ) of the CLT element. Using the tensile capacity ( $T$ ) of the hold-down, the neutral axis ( $x$ ) is  
 114 then determined by transitional equilibrium (Eq. 4) resulting in the expression for  $x$  (Eq. 5).

115 Tomasi [20] thus assumes that the foundation is infinitely stiff compared to the CLT element.  
 116 Ringhofer [21] and Schickhofer & Ringhofer [22] presented a similar “stress block” methods  
 117 adding the possibility to consider a deformable CLT flooring underneath of the shear wall.



$$C = (0.8 \cdot x) \cdot f_c \cdot t_{ef} \quad (3)$$

$$C - q \cdot w - T = 0 \quad (4)$$

$$x = \frac{q \cdot w + T}{0.8 \cdot f_c \cdot t_{ef}} \quad (5)$$

118 **Figure 5.** Illustration of the “stress block” method as proposed in [20].

119 The tension force in the hold-down (T) is then determined by means of rotational equilibrium  
 120 (Eq. 5) in the center of the panel:

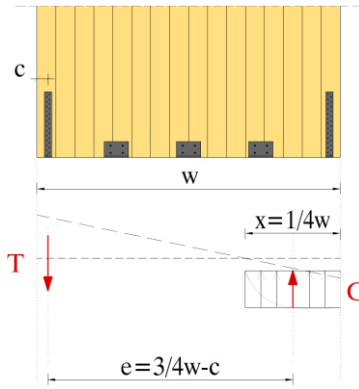
$$121 \quad -F_R \cdot h + T \cdot \left(\frac{w}{2} - c\right) + C \cdot \left(\frac{w}{2} - 0.4 \cdot x\right) = 0 \quad (6)$$

122 Using the expression in Eq. 3 for the resultant compressive force (C) and the expression in Eq.  
 123 5 for the neutral position (x), the total lateral force on the wall is then calculated as:

$$124 \quad F_R \cdot h = T \cdot \left(\frac{w}{2} - c\right) + (q \cdot w + T) \cdot \left(\frac{w}{2} - \frac{(q \cdot w + T)}{2 \cdot f_c \cdot t_{ef}}\right) \quad (7)$$

125 **Method C – Wallner-Novak et al. [15]**

126 Wallner-Novak et al. [15] proposed a similar method but with a different length of the  
 127 compression zone (x) corresponding to ¼ of the wall width (Fig. 6), and a 10 % reduced effect  
 128 of the vertical load (q) emanating from the partial safety factor for permanent loads. Rotational  
 129 equilibrium (Eq. 8) yields the expression (Eq. 10) for the total lateral force (F<sub>R</sub>):



$$T = \frac{F_R \cdot h}{e} - \frac{(0.9 \cdot q) \cdot w}{2} \quad (8)$$

$$e = \frac{3}{4} \cdot w - c \quad (9)$$

$$F_R \cdot h = \left( T + \frac{(0.9 \cdot q) \cdot w}{2} \right) \cdot \left( \frac{3}{4} \cdot w - c \right) \quad (10)$$

130 **Figure 6.** Illustration of the internal lever arm ( $e$ ) as proposed in [15].

131 In contrary to the general sliding resistance ( $F_T$ ) calculated as the sum of the resistance of the  
 132 angle brackets, Wallner-Novak et al. [15] included the contribution of friction (with a friction  
 133 coefficient  $\mu = 0.4$ ) of the vertical load ( $q$ ) to the sliding resistance of the shear wall (Eq. 11):

$$134 \quad F_T = \sum H_i + \mu \cdot (0.9 \cdot q) \cdot w \quad (11)$$

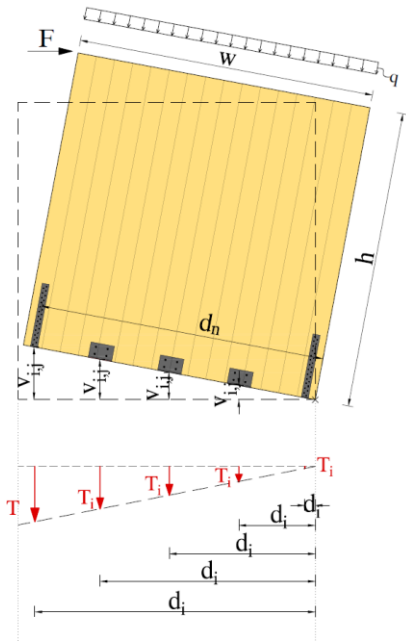
#### 135 **Method D – Pei et al. [23]**

136 Pei et al. [23] presented a method that considers the CLT panel as a rigid body rotating around  
 137 one of its corners (Fig. 7). Pei et al. [24], Shen et al. [25], Karakabeyli & Douglas [26] and  
 138 Gavric & Popovski [27] all presented similar methods. It should be specifically noted that the  
 139 proposed simplified kinematic method does not explicitly consider the sliding resistance of the  
 140 shear wall. Instead a connection resistance was “back-calibrated” by comparing the model  
 141 hysteretic obtained from numerical modelling with experimental measurements [23, 24] so that  
 142 the load-carrying capacity is limited by rigid body rotation around one of the panels corners.  
 143 Therefore, care should be taken when comparing this method to other similar methods.

144 To determine the lateral force, the connector’s elongation and stiffness/strength is considered.  
 145 The tensile strength of each connector is proportional with the distance of the connector from  
 146 the panel edge. A triangular distribution of the connector displacement is considered based on  
 147 that the furthest connector (the left hold-down according to Fig. 7) reaches its total elastic  
 148 tensile strength ( $T$ ). Imagining the remaining connections as elastic springs, they will elongate



149 based on a triangular distribution and thus their tensile strength is proportional with their  
 150 distance ( $d_i$ ) from the rotational point. The calculation steps for Method D are as follows:



1. Determine the tensile strength ( $T$ ) of the connector furthest from the point of rotation.
2. Calculate the elongation ( $v_{i,y}$ ) for the hold-down based on its vertical stiffness ( $k^v$ ) and capacity ( $T$ ).
3. Calculate the elongation ( $v_{i,y}$ ) for each connector based on a triangular distribution.
4. Calculate the tensile strength for each connector based on its stiffness ( $T_i = v_{i,y} \cdot k_i^V$ ).
5. Calculate the total rotational resistance in terms of the total lateral load ( $F_R$ ):

$$F_R \cdot h = \sum_{i=1}^n T_i \cdot d_i + \frac{q \cdot w}{2} \cdot d_n \quad (12)$$

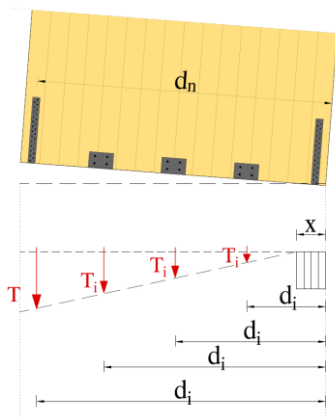
where  $d_n$  is the distance from the panel edge to the furthest connector, which typically is the hold-down.

151

152 **Figure 7.** Illustration of, and calculation steps for Method D based on [23].

153 **Method E – Reynolds et al. [18]**

154 Reynolds et al. [18] presented a method similar to Method D with a triangular distribution of  
 155 the tensile capacity but with the addition of a compressive zone (Fig. 8). The calculation steps  
 156 are; 1) determine the tensile strength ( $T$ ) of the connector furthest from the point of rotation, 2)  
 157 calculate the tensile capacity ( $T_i$ ) of remaining connectors based on a triangular distribution  
 158 (Eq. 13), 3) calculate the compression zone ( $x$ ) of the wall (Eq. 14), and 4) determine the lateral  
 159 resistance ( $F_R$ ) of the shear wall (Eq. 15).



$$T_i = T \cdot \frac{d_i}{d_n} \quad (13)$$

where  $d_n$  is the distance from the panel edge to the furthest connector, and where  $T_i$  should not exceed the maximum capacity of the actual connector

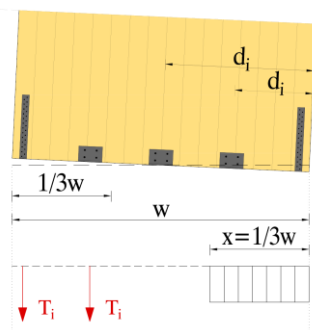
$$x = \frac{q \cdot w + \sum T_i}{f_c \cdot t_{ef}} \quad (14)$$

$$F_R \cdot h = \sum_{i=1}^n T_i \cdot \left(d_i - \frac{x}{2}\right) + \frac{q \cdot w^2}{2} - (q \cdot w) \cdot \frac{x}{2} \quad (15)$$

160 **Figure 8.** Triangular distribution of tensile capacity as proposed by [18].

161 **Method F – Reynolds et al. [18]**

162 Reynolds et al. [18] presented a method combining the kinematic equilibrium of Method D  
 163 with a compressive zone of  $\frac{1}{3}$  of the panel width, similar to Methods B and C. This method  
 164 only considers the resistance of the connectors placed in a “tensile zone” a distance of  $\frac{1}{3}$  of the  
 165 width from the panel edge (Fig. 9). Assuming that the resultant force from the vertical load is  
 166 centered in the panel, and defining the distance ( $d_i$ ) to each connector in the “tensile zone”, the  
 167 lateral load-carrying capacity ( $F_R$ ) can be calculated as:



$$F_R \cdot h = \sum_{i=1}^n T_i \cdot \left( d_i - \frac{w}{6} \right) + \frac{q \cdot w^2}{3} \quad (16)$$

$T_i$  – tensile strength of the connector furthest from the compression zone.

$$d_i - \frac{2 \cdot w}{3} < d_i < w.$$

168 **Figure 9.** Illustration of compression and “tensile” zones based on [18].

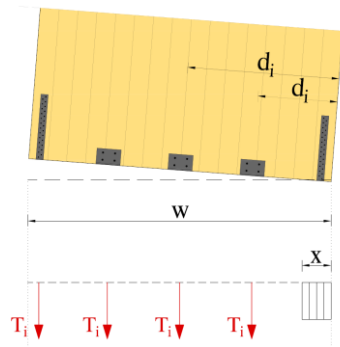
169 In this case, only the angle brackets outside of the tensile zone are available to resist sliding of  
 170 the wall. However, contrary to the general sliding resistance ( $F_T$ ) calculated as the sum of the  
 171 resistance of the angle brackets, Reynolds et al. [18] included the contribution of friction (with  
 172 a friction coefficient = 0.2) to the sliding resistance of the shear wall (Eq. 17):

$$F_T = \sum H_i + 0.2 \cdot (\sum T_i + q \cdot w) \quad (17)$$

174 **Method G – Reynolds et al. [18]**

175 Reynolds et al. [18] presented another method similar to Method F but with a reduced  
 176 compression zone (Eq. 18). In addition, the amount of connectors providing overturning  
 177 resistance is increased to encompass all connectors outside of the compression zone (Fig. 10)  
 178 with the exception that “any connectors required to resist sliding are excluded” [18]. The  
 179 tensile resistance of each connector is taken as their maximum elastic capacity. Using a simple

180 rotational equilibrium then gives the lateral resistance of the shear wall (Eq. 19). The sliding  
 181 resistance of the wall is calculated in the same manner as for Method F (see Eq. 17).



$$x = \frac{q \cdot w + \sum T_i}{f_c \cdot t_{ef}} \quad (18)$$

$\sum T_i$  – sum of vertical strength of the connectors activated in rotation

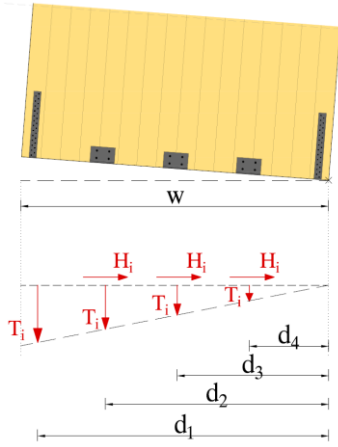
$$F_R \cdot h = \sum_{i=1}^n T_i \cdot \left( d_i - \frac{x}{2} \right) + \frac{q \cdot w^2}{2} - (q \cdot w) \cdot \frac{x}{2} \quad (19)$$

where  $d_i > x$  as only connectors within the tension zone are considered.

182 **Figure 10.** Suggested method with extended “tensile” zone based on [18].

183 **Method H – Gavric & Popovski [27]**

184 Gavric & Popovski [27] argued that the current proposed methods are too simplistic as they do  
 185 not consider the interaction of shear and tension forces in the connectors. The proposed method  
 186 considers interaction of shear and tension forces specifically in the angle brackets as tests  
 187 showed that hold-downs does not provide any significant resistance in the shear direction [27].  
 188 An iterative process (Fig. 11) was applied to calculate a so called unreduced factored wall  
 189 lateral resistance ( $F^*$ ) and then iteratively reducing the “real” lateral load ( $F$ ) until the  
 190 interaction (circular or triangular) of shear and tension forces in angle brackets are within its  
 191 limit. Rinaldin & Fragiacomio [28] analyzed the interaction domain and proposed a circular  
 192 interaction to the power of two, as the most appropriate, but for ease of calculation, this paper  
 193 will use the triangular verification of interaction (Eq. 20).



1. Calculate the resistance to sliding,  $F_T^*$  (Eq. 21)
2. Calculate the resistance to rotation  $F_R^*$  (Eq.22)
3. Specify  $F^* = \min\{F_T^*; F_R^*\}$
4. Assume reduced “real” resistance  $F < F^*$
5. Calculate reduced  $H_i^*$  and  $T_{i+1}^*$
6. Iterate until interaction of shear and tension in the most loaded angle bracket is verified.

$$\frac{H_i^*}{H_i} + \frac{T_{i+1}^*}{T_i} \leq 1 \quad (20)$$

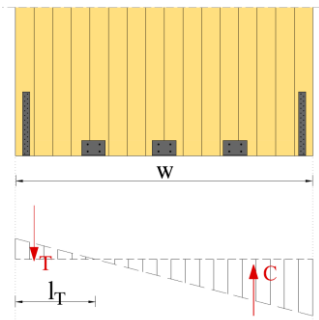
194 **Figure 11.** Suggested iterative method as suggested by [27].

$$195 \quad F_T^* = n_{AB} \cdot H_i \quad (21)$$

$$196 \quad F_R^* = \left( \frac{q \cdot w^2}{2 \cdot h} \right) + \frac{T_1 \cdot d_1}{h} + \frac{T_{i+1}}{d_1 \cdot h} \cdot \sum (d_{i+1})^2 \quad (22)$$

197 **Method I – Schickhofer et al. [29]**

198 Schickhofer et al. [29] presented a theoretical method (Fig 12) assuming a linear elastic and  
 199 continuous behavior of the bottom joint [30]. Distributing the overturning moment from the  
 200 lateral load ( $F_R$ ) and including the contributing of the vertical load ( $q$ ) makes it possible to  
 201 evaluate the length of the tensile zone ( $l_T$ ) (Eq. 24) and the tensile force ( $T$ ) in the hold-down  
 202 by equilibrium equations (Eq. 25). Using the maximum tensile capacity of the hold-down, the  
 203 lateral resistance ( $F$ ) can then be calculated.



$T$  – tensile load in hold down connector

$$T = \left( \frac{6 \cdot F_R \cdot h}{w^2} - q \right) \cdot \frac{l_T}{2} \quad (23)$$

where  $l_T$  is the length of the tension zone

$$l_T = \frac{1}{2} - \frac{q \cdot w^3}{12 \cdot F_R \cdot h} \quad (24)$$

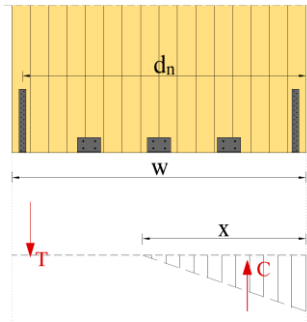
$$T = \frac{3 \cdot F_R \cdot h - q \cdot w^2}{2 \cdot w} + \frac{q^2 \cdot w^3}{24 \cdot F_R \cdot h} \quad (25)$$

204 **Figure 12.** Suggested theoretical method based on [29].

205 **Method J – Schickhofer et al. 2010 [29]**

206 Schickhofer et al. [29] presented a method combining a triangular compression zone with  
 207 tensile bracing (Fig 13), depicting a situation where the lateral force is just large enough to

208 cause the wall to rotate. The problem has three unknowns; the length of the compression zone  
 209 ( $x$ ), the maximum compressive force at the corner of the panel ( $N_c$ ) and the load in the tensile  
 210 bracing ( $T$ ). However, with only two equilibrium equations being available, two solutions to  
 211 the problem was proposed. The first solution (Eq. 26) assumes that the tensile bracing reaches  
 212 its ultimate elastic capacity ( $T$ ) in which case the lateral load ( $F_{R,I}$ ) is limited by the maximum  
 213 compressive stress at the corner of the panel ( $N_c \leq f_c \cdot A_{ef}$ ) where  $A_{ef}$  is the effective area per  
 214 length  $m$  shear wall, i.e.,  $A_{ef} = t_{ef} \cdot l$ .



**Solution I:** Solve for  $F_{R,I}$  by limiting  $N_c$  to  $f_c \cdot A_{ef}$

$$N_c = \frac{4 \cdot (q \cdot w)^2 + 8 \cdot q \cdot w \cdot T + 4 \cdot T^2}{3 \cdot w \cdot (q \cdot w + T) - 6 \cdot (F_{R,I} \cdot h - T \cdot d_n)} \quad (26)$$

**Solution II:** Solve for  $F_{R,II}$  by limiting  $T$  to the capacity of the connector furthest from the compression edge:

$$T = \frac{1}{8} \cdot [-8 \cdot (q \cdot w) + N_c \cdot (3 \cdot w + 6 \cdot e)] - \frac{\sqrt{3}}{8} \cdot \sqrt{N_c \cdot [N_c \cdot (3 \cdot w^2 + 12 \cdot w \cdot e + 12 \cdot e^2) - (32 \cdot ((F_{R,II} \cdot h) + (q \cdot w \cdot e)))]} \quad (27)$$

215 **Figure 13.** Suggested method based on [29].

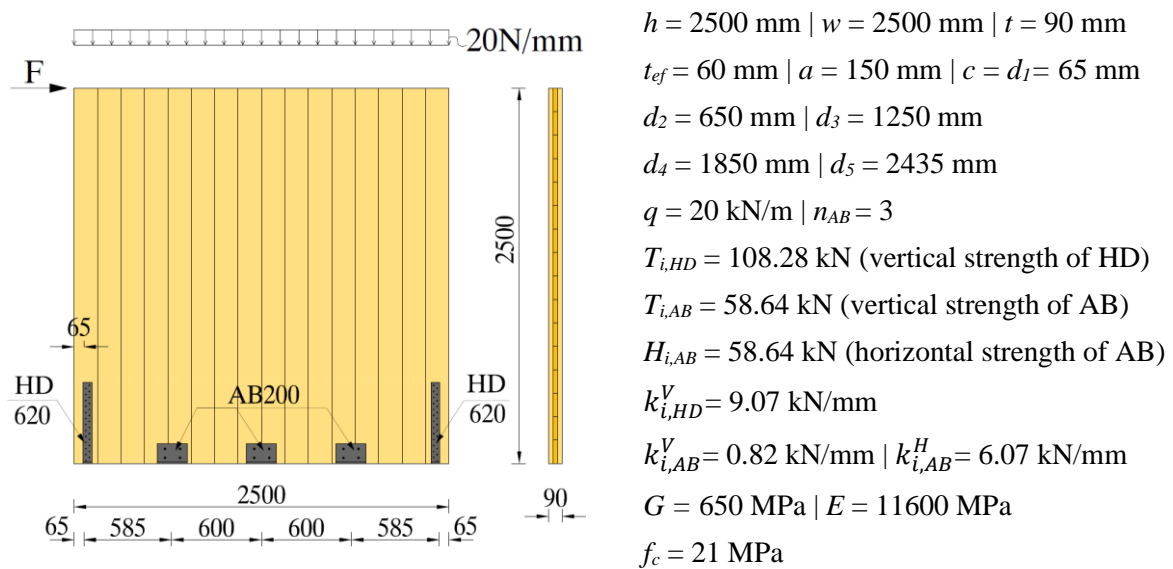
216 The second solution (Eq. 27) assumes that the corner of the panel (Fig. 13) reaches its ultimate  
 217 compressive capacity ( $f_c$ ) in which case the lateral load ( $F_{R,II}$ ) by limiting ( $T$ ) to the tensile  
 218 capacity of tensile bracing. In Eq. 27, the lever arm  $e$  is calculated as  $e = d_n - w/2$ . The lateral  
 219 capacity of the shear wall with respect to rotation is then evaluated as  $F_R = \min(F_{R,I}; F_{R,II})$ .  
 220 Tamagone et al. [31] proposed a similar method utilizing the same failure modes but instead  
 221 an iterative process was proposed to calculate the reaction force in the connections by varying  
 222 the position of the natural axis. As this method requires the use of Finite Element software for  
 223 its calibration, it is out of scope of this paper.

### 2.3. Comparison of strength methods

225 A calculation example is used to provide an indication of differences between the presented  
 226 analytical methods. The comparison is based on a standard shear wall setup (Fig. 14) anchored  
 227 to a concrete foundation with hold-downs (HD) and angle brackets (AB). The connectors used

228 in this example are named HD620 and AB200, which is analogous with the connectors tested  
 229 in Tomasi [32] and Tomasi & Smith [33]. For comparison, a 3-layered square (2500x2500 mm)  
 230 CLT panel with a total thickness of 90 mm was used. The connections are placed on one side  
 231 of the shear wall only. Maximum values based on tests were used for the connectors' tensile  
 232 strength and stiffness while the characteristic value of the compressive strength parallel to the  
 233 grain was used for the CLT panel.

234 Test results for the example shear wall is presented in [32, 34]. Test result for the vertical  
 235 strength of the angle bracket is not available, but according to Gavric & Popovski [27] it can  
 236 be assumed that the vertical capacity equals its horizontal capacity, which is also supported by  
 237 test results [33]. Method D includes the vertical stiffness of the angle brackets for which no  
 238 test data was found. In this case, a vertical stiffness value of a softer angle bracket was used  
 239 which coincide with other tested connectors [32, 33].



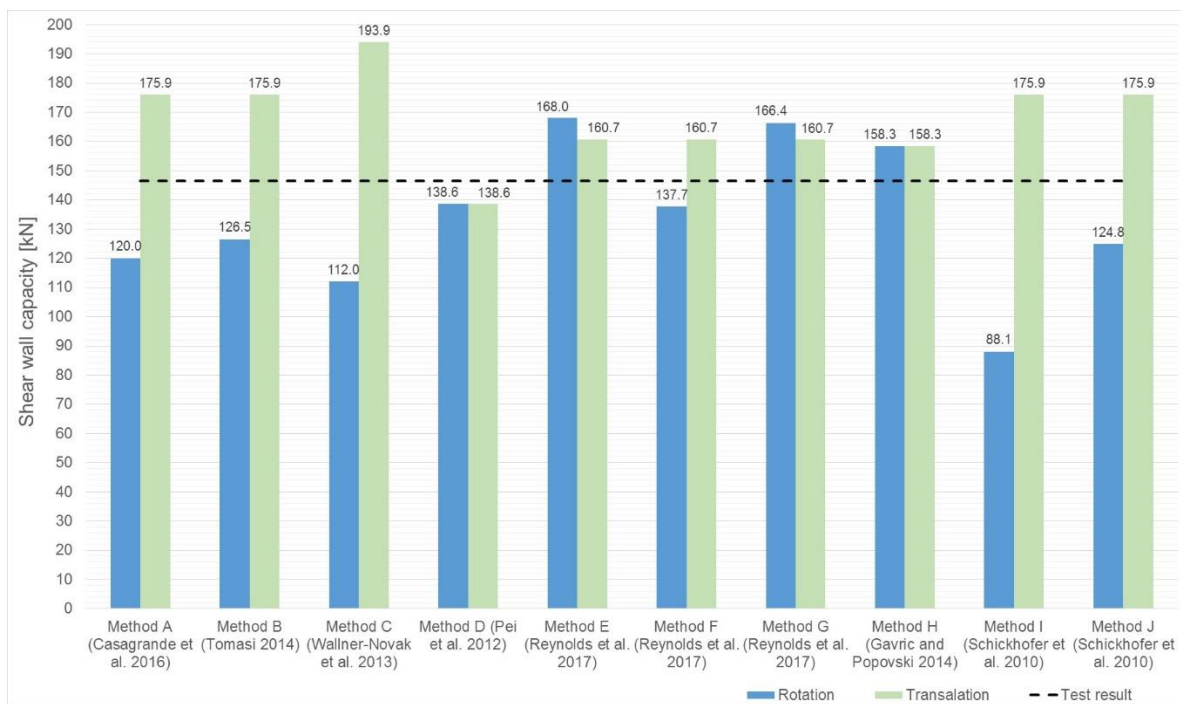
240

241 **Figure 14.** Example shear wall with geometry and material data.

242 Based on data presented in Fig. 14, the lateral capacity in the tension and shear directions for  
 243 the 10 different methods are illustrated in Fig. 15. The example wall has strong angle brackets  
 244 in the shear direction which means that the load-carrying capacity of Methods A, B, C, I and J  
 245 in translation is much higher than the load-carrying capacity for the respective wall in rotation.

246 Methods D represents a special case where an unmodified tensile strength based on tests was  
 247 used to evaluate the wall capacity which is not what was suggested by [27]. However, at least  
 248 for this wall setup, this approximation provides adequate results.

249 For Method H, the lateral load-capacity is evaluated based on a triangular interaction of the  
 250 resistance in tension and shear. The Methods C, E, F and G all consider the increased shear  
 251 capacity due to friction. Methods E, F, G represents a special case as the lateral load-carrying  
 252 capacity is evaluated from the connectors providing overturning resistance that are not  
 253 necessary to resist sliding [18]. For the considered wall setup, two angle brackets are required  
 254 to resist sliding in Methods E and F, which means that the vertical capacity of one angle  
 255 brackets is used to increase the capacity in rotation.



256

257 **Figure 15.** Rotational and shear capacities calculated from the 10 strength methods.

258 **3. STIFFNESS OF CLT SHEAR WALLS**

259 The force-displacement relation determines the stiffness of a shear wall. By knowing the  
 260 maximum force acting on the shear wall, and the stiffness of the connectors, the wall  
 261 displacement can be calculated. Analytical methods for the displacement of CLT shear walls

262 are based on the different contributions to shear wall deformation (Fig. 2); translational  $\Delta_T$  (or  
 263 slip), rotational  $\Delta_R$  (or rocking), panel shear  $\Delta_S$  and panel bending  $\Delta_B$ . Due to the relatively high  
 264 in-plane stiffness of the CLT element, the rocking mechanism is generally dominant but  
 265 different shear wall geometries, hold-downs and angle brackets with different strength and  
 266 stiffness characteristics can have substantial effect [15, 16, 17].

### 267 **3.1. Methods for stiffness assessment**

268 In the literature, five methods for assessing the stiffness of CLT shear wall are identified;  
 269 Casagrande et al. [19], Hummel et al. [35], Wallner-Novak et al. [15], Gavric et al. [36],  
 270 Fletcher & Schickhofer [37]. If not otherwise stated, the state-of-the-art use the generalized  
 271 notations previously presented in Fig. 3 and the total wall displacement is calculated as:

$$272 \Delta_{TOT} = \Delta_B + \Delta_S + \Delta_T + \Delta_R \quad (29)$$

#### 273 **Method I – Casagrande et al. [19]**

274 Casagrande et al. [19] considered the contribution of the in-plane shear deformation ( $\Delta_S$ ), rigid-  
 275 body translation ( $\Delta_T$ ) and rigid-body rotation ( $\Delta_R$ ) which is analogous to simplifications made  
 276 by, e.g. Vessby [38] and Reynolds et al [39]. Similar to the assessment of the load-carrying  
 277 capacity (presented as Method A above), Method I applies a level arm of 90 % of the width of  
 278 the panel to calculate the rocking deformation. The different contributions to shear wall  
 279 deformation are calculated as:

$$280 \Delta_T = \frac{F}{k_{AB}^H \cdot n_{AB}} \quad (30)$$

$$281 \Delta_S = \frac{F \cdot h}{G \cdot t \cdot w} \quad (31)$$

$$282 \Delta_R = \left( \frac{F \cdot h}{(0.9 \cdot w)} - \frac{q \cdot w}{2} \right) \cdot \frac{h}{k_{HD}^V \cdot (0.9 \cdot w)} \quad (32)$$

#### 283 **Method II – Hummel et al. [35]**

284 Besides the shear deformation of the CLT panel, Method II also considers rotation/rocking of  
 285 the wall panel due to tensile anchoring and contact, and slip of the wall panel due to shear  
 286 anchoring. Method II also considers the bending deformation of the CLT panel (Fig. 2). The



287 same contributions of deformations are also presented in Hummel [16], Seim et al. [40], and  
 288 Hummel & Seim [41].

289 For the shear deformation, a reduced effective shear modulus of the CLT wall panel is  
 290 considered. Hummel et al. [35] also considers the increased panel flexibility occurring from an  
 291 elastic foundation where two cases are considered; 1) a rigid foundation (e.g., a concrete slab),  
 292 and 2) an elastic foundation (e.g., a timber floor between stories with an elastic intermediate  
 293 layer). The different contributions to shear wall deformation are calculated as:

$$294 \quad \Delta_B = \frac{F \cdot h}{3 \cdot EI_{ef}} \quad (33)$$

$$295 \quad \Delta_S = \frac{F \cdot h}{GA_{ef}} \quad (34)$$

$$296 \quad \Delta_R = \begin{cases} \frac{h}{(w-2 \cdot c)} \cdot \frac{\max\left\{F \cdot \frac{h}{(w-2 \cdot c)} - \frac{q \cdot w}{2}; 0\right\}}{k_{HD}^v} - \text{rigid foundation} \\ \frac{h^2}{d_i - l_c/3} \cdot \frac{2 \cdot F}{k_D \cdot l_c^2} - \text{elastic foundation} \end{cases} \quad (35)$$

$$297 \quad \Delta_T = \frac{F}{n_{AB} \cdot k_{AB}^h} \quad (36)$$

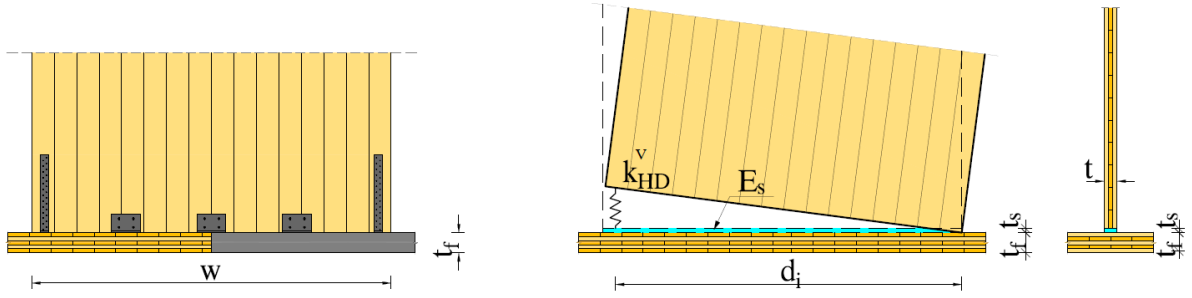
298 The flexural stiffness is determined based on Eq. 37, where  $t_{ef}$  is the thickness of the vertical  
 299 layers and  $w$  is the width of the CLT wall panel.

$$300 \quad EI_{ef} = E_0 \cdot \left[ \frac{t_{ef} \cdot w^3}{12} \right] \quad (37)$$

301 The shear stiffness is determined based on an effective shear modulus,  $G_{eff}$ , and the gross shear  
 302 area,  $A$ , where  $a$  is the average width of the lamellae. Based on the thickness of the lamella the  
 303 width  $a$  can vary between 80 and 240 mm, see for example [42]. The effective shear modulus  
 304 was derived by Schickhofer et al. [29]:

$$305 \quad GA_{ef} = G_{eff} \cdot A = \frac{G}{1 + 6 \cdot \left[ 0.32 \cdot \left( \frac{t}{a} \right)^{-0.77} \right] \cdot \left( \frac{t}{a} \right)^2} \cdot A, \text{ where } A = t \cdot w \quad (38)$$

306 A typical CLT shear wall with rigid/elastic is shown in Fig. 16. The rocking deformation can  
 307 be calculated as presented in Eq. 35 for both foundation types.



308

309 **Figure 16.** CLT wall with rigid foundation (left) and elastic intermediate layer (right).

310 Illustrations based on Hummel et al. [35].

311 In the case of the elastic foundation, the width of the elastic intermediate layer ( $b_s$ ), the E-  
 312 modulus of the layer ( $E_s$ ) and the length of pressure zone ( $l_c$ ) is required. Two cases are  
 313 distinguished for the width ( $b_s$ ), one for the case of an exterior wall and one for an interior wall,  
 314 (Eq. 39a,b) where  $t_f$  is the thickness of the floor element ( Fig. 16). The E-modulus ( $E_s$ ) can,  
 315 for example, be for the elastic material Sylodyn, a common elastic intermediate layer used in  
 316 CLT walls systems. The use of the elastic intermediate layer contributes to an increased rocking  
 317 deformation due to the reduced stiffness ( $k_D$ ) of the elastic foundation (Eq. 40).

318 
$$b_s = t + \frac{1}{4}t_f \text{ – for exterior wall} \quad (39a)$$

319 
$$b_s = t + \frac{1}{2}t_f \text{ – for interior wall} \quad (39b)$$

320 
$$k_D = \frac{E_s \cdot b_s}{t_s} \quad (40)$$

321 **Method III – Wallner-Novak et al. [15]**

322 Method III is presented by Wallner-Novak et al. [15] which considers the same contributions  
 323 as Method II, with only a slight difference in the definition of the shear stiffness of the CLT  
 324 panel due to a reduced shear modulus. The total displacement is calculated based on panel  
 325 bending and shear as well as contributions from translation and rocking deformation:

326 
$$\Delta_B = \frac{F \cdot h^3}{3 \cdot EI_{ef}} \quad (41)$$

327 
$$\Delta_S = \frac{F \cdot h}{GA} \quad (42)$$

328 
$$\Delta_T = \frac{F}{n_{AB} \cdot k_{AB}^H} \quad (43)$$

$$\Delta_R = \left[ \frac{F \cdot h}{w} - \frac{q \cdot w}{2} \right] \cdot \frac{h}{w \cdot k_{HD}^V} \quad (44)$$

The bending stiffness ( $EI_{ef}$ ) used in Eq. 41 is calculated according to Eq. 37 while the shear stiffness used in Eq. 42 is determined using a 25% reduction of the shear modulus:

$$GA = (0.75 \cdot G) \cdot (t \cdot w) \quad (45)$$

#### Method IV – Gavric et al. [36]

This method was originally presented by Gavric et al. [43]. Gavric et al. [36], who argued that previous methods fail to take into account the tensile characteristics of angle brackets, proposed a method that takes into account all the stiffness and strength components of hold-downs and angle brackets also in their weaker directions. By introducing a vertical stiffness of angle brackets, a friction coefficient to reduce sliding, and a shape reduction factor of 1.2 for the shear deformations, the deformations can be calculated as:

$$\Delta_B = \frac{F \cdot h^3}{3 \cdot EI_{ef}} \quad (46)$$

$$\Delta_S = \frac{1.2 \cdot F \cdot h}{GA_{ef}} \quad (47)$$

$$\Delta_T = \frac{F - \mu \cdot q \cdot w}{n_{AB} \cdot k_{AB}^H} \quad (48)$$

$$\Delta_R = \frac{\left( F \cdot h - q \cdot \frac{w^2}{2} \right) \cdot h}{\sum k_{HD}^V \cdot d_i^2 + \sum k_{AB}^V \cdot d_i^2} \quad (49)$$

As in Method III, here also, the bending stiffness ( $EI_{ef}$ ) in the bending deformation is calculated according to Eq. 37. According to Gavric et al. [36], the shear stiffness ( $GA_{ef}$ ), is calculated with a shear modulus ( $G$ ) equal to 0.69 GPa, and an effective shear area ( $A_{ef}$ ), which is considering just the vertical layers, is calculated as:

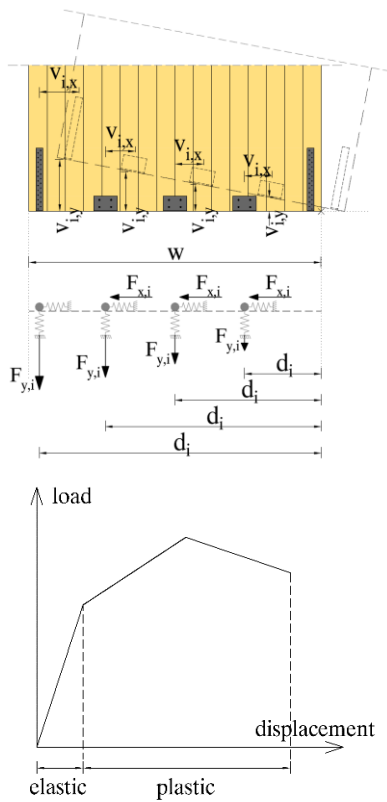
$$A_{ef} = t_{ef} \cdot w \quad (50)$$

In translation, a friction coefficient ( $\mu$ ) of 0.3 is used. In rotation the panel is considered rigid, and rotating around a corner of the wall, similar to the strength Method D. However, in calculating the stiffness properties of the connectors, Gavric et al. [36, 43] also considers the non-linear behavior of the force-displacement curve, suggesting that the stiffness of each

353 connector is evaluated based on the actual deformation of each connector. For this purpose,  
 354 three different stiffness ranges were proposed (compare Fig. 17); with an initial elastic stiffness,  
 355 a plastic stiffness until maximum load, and a negative stiffness phase until connection failure.  
 356 The sum of these evaluated stiffnesses at certain deformation intervals where then used to  
 357 calculate the rotation deformation (Eq. 49).

358 **Method V – Flatscher & Schickhofer [37]**

359 Flatscher & Schickhofer [37], and Flatscher [30] proposed a new displacement-based  
 360 calculation method for predicting the total load-displacement behavior of a CLT shear wall.  
 361 The fundamental difference of this method compared to force-based methods is that the sliding  
 362 and rocking behavior cannot be analyzed separately. Similar to the methods described in Gavric  
 363 et al. [36, 43] and Pei et al. [23], a rigid CLT body was assumed with a point of rotation at the  
 364 lower corner of the wall element to predict the behavior of the connections' (Fig. 17).



1. Assume a ratio  $p$  for the contribution of  $\Delta_T$  and  $\Delta_R$  to the total connection based deformation  $v_{con}$ .
2. Calculate the deformation of each connector in the shear ( $v_{i,x}$ ) and tensile directions ( $v_{i,y}$ ).

$$v_{i,x} = \Delta_T = p \cdot v_{con} \quad (51)$$

$$v_{i,y} = d_i \cdot \frac{(1-p) \cdot v_{con}}{h} \quad (52)$$

3. Evaluate a force  $F_{x,i}$  and  $F_{y,i}$  of each connector from their respective load-deformation relation.

$$F_{x,i} = f(v_i^x) = k_i^H \cdot v_{i,x} \quad (53)$$

$$F_{y,i} = f(v_i^y) = k_i^V \cdot v_{i,y} \quad (54)$$

4. Calculate the total lateral load on the wall based on sliding ( $F_T$ ) and rocking ( $F_R$ ).

$$F_T = \sum F_{x,i} + (\sum F_{y,i} + q \cdot w) \cdot \mu \quad (55)$$

$$F_R = \frac{1}{h} \cdot \left[ \sum (F_{y,i} \cdot d_i) + \frac{q \cdot w^2}{2} \right] \quad (56)$$

5. As only one lateral force can be active at a time, iterate until  $F_T = F_R$  by changing the ratio  $p$ .

365 **Figure 17.** Proposed displacement-based method showing the wall setup (based on [37, 30])

366 and a schematic force-displacement curve for a connector (inspired by [36]).

367 Evaluation of the maximum lateral force based on either sliding or rotation was made through  
 368 an iterative process (Fig. 17). Flatscher [30] proposed that the strength and deformation  
 369 characteristics of the connections is evaluated based on a multi-linear approximation of the  
 370 load-deformation curve taking into account the plasticization of connectors (Fig. 17). Finally  
 371 the contributions to deformation from panel shear and bending are calculated as

$$372 \quad \Delta_S = \frac{F \cdot h}{G^* \cdot w \cdot t} \quad (57)$$

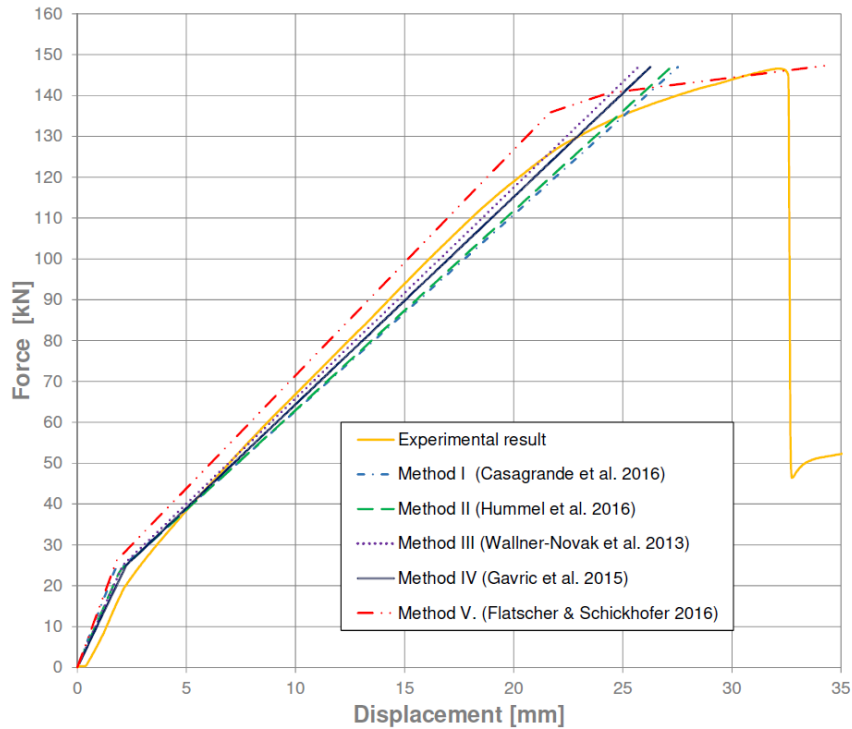
$$373 \quad \Delta_B = \frac{4 \cdot F \cdot h^3}{E_0 \cdot w^3 \cdot t_{ef}} \quad (58)$$

374 The shear contribution is depending on an effective shear modulus,  $G^*$  (Eq. 59) based on [1],  
 375 where  $p_s$  depends on the number of layers (0.53 for a 3 layered CLT element and 0.43 for a 5  
 376 layered CLT element).

$$377 \quad G^* = \frac{G}{1 + 6 \cdot p_s \cdot \left(\frac{t}{a}\right)^{1.21}} \quad (59)$$

### 378 **3.2. Comparison of stiffness methods**

379 The stiffness methods are compared with each other using the load-deformation relation of the  
 380 wall setup presented in Fig. 14 with an ultimate lateral resistance of 146.6 kN [32, 34]. The  
 381 total displacement for each method is calculated as the sum of the displacement mechanisms  
 382 (Fig. 2), bending, shear, translation and rotation according to Eq. 29. In Fig. 18, the results  
 383 obtained for each method are illustrated as a load-deformation behavior and as a contribution  
 384 by each deformation mechanism at ultimate load. The initial higher elastic stiffness of Fig. 18  
 385 can be explained by the positive contribution of the vertical load. When the lateral force is large  
 386 enough to cause the wall to rotate, the positive impact of the vertical load is lost because the  
 387 hold-down is activated and the stiffness of the wall is reduced. This contribution has been taken  
 388 into consideration for each method so as to get a better correlation with the test results.



389

390 **Figure 18.** Comparison of test result with calculated load-deformation behavior.

391

From Fig. 18, it seems that for this specific shear wall configuration, all of the methods are able

392

to predict the elastic behavior with Method V slightly overestimating, and Methods I, II and

393

IV slightly underestimate the elastic stiffness. The method that most accurately seem to predict

394

the elastic stiffness is Method III. It should be noted that for Methods IV and V, a linear elastic

395

load-deformation behavior was assumed for both the hold-downs and the angle brackets, which

396

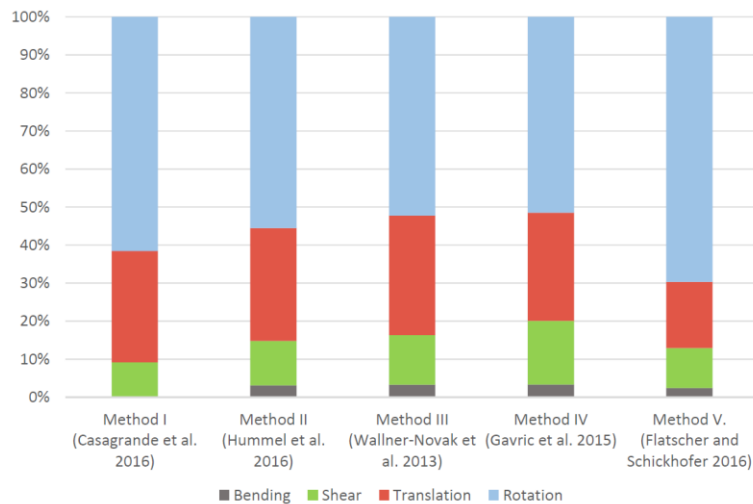
was not proposed by the respective authors. The reason why Method V is able to predict a

397

second linear stiffness is because the iteration of the deformation contribution of the

398

connections utilizes the strength of each connector which cannot exceed its capacity.



399

400 **Figure 19.** Illustration of displacement contribution for each method at the ultimate lateral load.

401

For this specific shear wall configuration, the contribution of the bending deformations to the total deformation of the shear wall is low (Fig. 19) which is due to the high flexural stiffness of the CLT material in relation to the stiffness of the connectors. Also, in this case the panel shear deformation are substantially lower than the connection based deformations.

405

#### 4. DISCUSSION AND CONCLUDING REMARKS

406

This paper presents and discuss the methods used to assess the behavior of shear walls in term of strength and stiffness. The methods proposed in the state-of-the-art are in general quite simple, utilizing static and kinematic equilibrium based on rigid body rotation of the CLT panel to evaluate the forces in connectors. The CLT shear wall is in this paper is viewed as a single wall panel without any openings and without any connections to other vertical or horizontal panels. The methods found in the literature are exclusively based on hold-down connector and angle brackets that are mainly used to resist rotation and sliding respectively, which is the current state of practice in CLT construction. The applicability of these methods is strongly dependent on the connection system and further study is required to include other possible connection systems. The redundancy present in real structures was purposely neglected in this overview with the aim to validate simple design approaches.

417

##### 4.1. Strength methods

418 This paper presents 10 different methods used to assess the lateral strength of CLT shear walls  
419 which is typically limited by either the lateral resistance in rotation or sliding. For the case  
420 study shear wall used in this paper, the lateral resistance based on rotation governs the behavior  
421 of the shear wall. However, it should be especially be noted that different shear wall  
422 configuration and different connection systems can have a substantial effect on the behavior of  
423 the shear wall. Therefore, the calculations presented in this paper should merely be seen as an  
424 example and a thorough study of different shear wall configurations is required in order to  
425 evaluate the analytical approach that best approximates the real shear wall behavior. The  
426 strength methods can be divided into two main groups in regards to their application:

427 1) *Methods A, B, C, I and J* only consider an internal lever arm between the tensile bracing  
428 and the compression zone which length mainly vary from the size of the compression zone.

429 To resist rotation of the shear wall, only the connector furthest from the point of rotation  
430 is considered while the angle brackets are designed to exclusively resist sliding. These  
431 methods typically consider only a few variables that can be readily determined from test  
432 results or based on producer data. The simplistic use of these methods, enables quick  
433 assessment of the lateral strength of a CLT shear wall.

434 2) *Methods D, E, F, G, and H* does, in addition to an internal lever arm, also consider the  
435 vertical capacity of the shear connectors. Even though their application is slightly more  
436 complex, these methods are still straightforward to use and seem to more accurately model  
437 the real behavior of the CLT shear wall. The vertical strength of angle brackets must be  
438 defined, information which in some cases can be difficult to obtain as these connectors are  
439 typically used only to resist sliding.

#### 440 **4.2. Stiffness methods**

441 To assess the displacement of a shear wall requires the calculation of shear and bending  
442 deformations in the CLT panel itself and the panel rocking and sliding behavior that is



443 dependent on the stiffness of the connectors used. This paper presents five methods to assess  
444 the displacement (stiffness) of CLT shear walls. Method I is neglecting the bending  
445 deformation of the CLT panel, while the other methods consider bending even though its  
446 contribution to the total shear wall deformation is typically low. As expected, the majority of  
447 the stiffness of the CLT shear wall relies on the stiffness of the connectors themselves. The  
448 methods are quite comparable, providing similar results. However, if the connection system is  
449 changed, the results from the different contributions seem to change significantly, indicating  
450 that the models are sensitive to the vertical and horizontal stiffness of the shear connectors.  
451 Similar to what was observed for strength assessment, the methods proposed are exemplified  
452 using one single shear wall configurations that does not fully describe the redundancy of a real  
453 structure. It is worth saying that in many practical applications it is more relevant to assess the  
454 relative stiffness of components rather than obtaining accurate results. Therefore, the  
455 usefulness of an analytical method should be related to its ability to correctly describe the  
456 stiffness as influenced by, for example, the vertical load and the connector stiffness.

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